THE

# THETA-PHI DIAGRAM

PRACTICALLY APPLIED TO

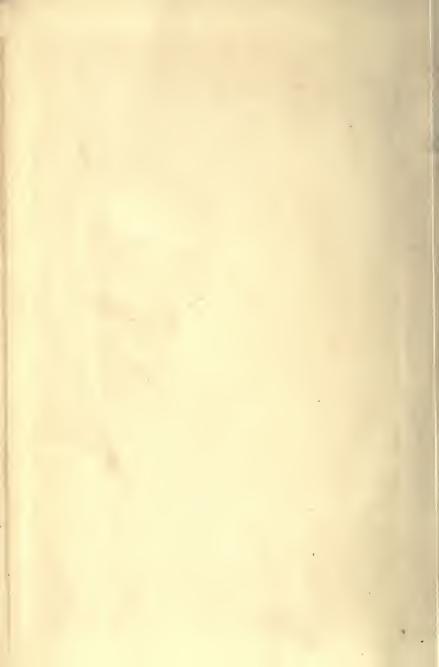
STEAM, GAS, OIL AND AIR ENGINES.

BY

HENRY A. GOLDING, A.M.I.M.E.







KEN CO

THE

## THETA-PHI DIAGRAM

PRACTICALLY APPLIED TO

STEAM, GAS, OIL, AND AIR ENGINES.

BY

### HENRY A. GOLDING, A.M.I.M.E.,

Chief Draughtsman, Messrs. B. Donkin and Co., and Assistant Lecturer, South-Western Polytechnic. London.

PRICE THREE SHILLINGS NET.

1898.

THE TECHNICAL PUBLISHING CO. LIMITED, 31, WHITWORTH STREET, MANCHESTER.

JOHN HEYWOOD,

29 AND 30, SHOE LANE, LONDON; AND RIDGEFIELD, MANCRESTER,
And all Booksellers.

69299

Digitized by the Internet Archive in 2007 with funding from Microsoft Corporation

#### PREFACE.

In the following pages an attempt has been made to present in as simple and practical a manner as possible, the use of the temperature-entropy diagram and the various methods of drawing it for different heat motors. That the subject presented peculiar difficulties, because of its unfitness for presentation in a popular manner, will readily be granted; but I venture to think that one of the principal reasons for the lack of knowledge upon the subject by draughtsmen, steam students, and others has been the want of an elementary work, not overcrowded with mathematics. Most of the literature upon the subject has presented the mathematical rather than the graphical side of the question, with the result that students have become afraid of tackling what they believe to be an intricate mathematical investigation.

Of the utility of the temperature-entropy diagram in representing the various thermal changes which take place in all heat motors there cannot be any doubt. To quote only one authority, Mr. Mark H. Robinson, in the discussion on Mr. Willans' last paper, said: "Up to a certain point the practical man might ignore the present paper, and others like it; but if he aspired to design economical steam engines, he might derive more good from the study of, say, Mr. Maefarlanc Gray's  $\theta \phi$  diagram than from many portfolios of working drawings."

Where authorities have been quoted or made use of, the particulars are given in the text, but I will take this opportunity of expressing my indebtedness to Professor Ewing for his work on "The Steam Engine and other Heat Engines," and his Cantor Lectures on the "Mechanical Production of Cold"; to Professor Boulvin, for his articles in La Revue de Mecanique; and to various papers, principally those by the late Mr. P. W. Willans and Mr. Macfarlane Gray, published in the Proceedings of the Institutions of Civil and Mechanical Engineers. I also wish to thank the Council of the latter Institution for permission to reproduce some of the indicator diagrams and figures given in the reports of the Steam Jacket Research Committee.

I regret that in Chapter IV., page 70, a slight inaccuracy should have occurred. Referring to the late Mr. Willans' method of calculating the thermal efficiency in his central valve engine tests, I find that the lower temperature of 110 deg. Fah. was only assumed for the preliminary calculations of the comparative losses due to incomplete expansion in condensing and non-condensing engines, but for the actual trials the temperature in the exhaust chamber was taken in each trial separately, and used for calculating the thermal efficiency. (See line 10 in Appendix, Table I., Willans on "Condensing Steam Engine Trials.") I am also indebted to Captain Sankey for pointing out that the specific heat of gases is not constant at the high temperatures occurring in gas and oil engines, and therefore the calculations which involve the use of Cp and Cv for a gaseous mixture at high temperatures must necessarily be looked upon as approximate only.

HENRY A. GOLDING.

London, September, 1898.

### CONTENTS.

	P	AGE
IN	PRODUCTION	vil.
	CHAPTER I ENTROPY.	
	TICLE	
	Introduction	1
	Entropy Diagrams	2
	Entropy	4
4.	Theoretical Entropy Diagram for Steam	5
	CHAPTER II.—Entropy of Water and Sigam.	
5.	Method of Constructing the Curves	7
в.	Entropy of Water	10
	Entropy of Steam	11
8.	Entropy Diagram for Ice, Water, and Steam	11
9.	Constant Volume Curves	15
	CHAPTER III.—Conversion of Indicator Diagram to Entropy Diagram.	
10.	Calculation of Dryness Fraction	21
10. 11.	Calculation of Dryness Fraction	21 23
10. 11. 12.	Calculation of Dryness Fraction	21 23 28
10. 11. 12. 13.	Calculation of Dryness Fraction	21 23
10. 11. 12. 13.	Calculation of Dryness Fraction Actual Example, Compound Engine Complete Entropy Diagram (Professor Boulvin's Chart). Application of Complete Entropy Diagram	21 23 28 33
10. 11. 12. 13. 14.	Calculation of Dryness Fraction	21 23 28 33 34
10. 11. 12. 13. 14.	Calculation of Dryness Fraction . Actual Example, Compound Engine . Complete Entropy Diagram (Professor Boulvin's Chart) Application of Complete Entropy Diagram	21 23 28 33 34 35
10. 11. 12. 13. 14.	Calculation of Dryness Fraction . Actual Example, Compound Engine . Complete Entropy Diagram (Professor Boulvin's Chart) Application of Complete Entropy Diagram	21 23 28 33 34 35
10. 11. 12. 13. 14. 15.	Calculation of Dryness Fraction Actual Example, Compound Engine Complete Entropy Diagram (Professor Boulvin's Chart).  Application of Complete Entropy Diagram θ φ Diagram for Carnot Cycle Condensation during Adiabatic Expansion Adiabatic Expansion of Wet Steam  CHAPTER IV.—Heat Losses.	21 23 28 33 34 35 37
10. 11. 12. 13. 14. 15. 16.	Calculation of Dryness Fraction	21 23 28 33 34 35
10. 11. 12. 13. 14. 15. 16.	Calculation of Dryness Fraction Actual Example, Compound Engine Complete Entropy Diagram (Professor Boulvin's Chart).  Application of Complete Entropy Diagram θ φ Diagram for Carnot Cycle Condensation during Adiabatic Expansion Adiabatic Expansion of Wet Steam  CHAPTER IV.—Heat Losses.	21 23 28 33 34 35 37
10. 11. 12. 13. 14. 15. 16.	Calculation of Dryness Fraction Actual Example, Compound Engine Complete Entropy Diagram (Professor Boulvin's Chart).  Application of Complete Entropy Diagram θ φ Diagram for Carnot Cycle Condensation during Adiabatic Expansion Adiabatic Expansion of Wet Steam.  CHAPTER IV.—Heat Losses.  Effect of Steam Jacketing Theoretical Entropy Diagram for Superheated Steam	21 23 28 33 34 35 37
10. 11. 12. 13. 14. 15. 16.	Calculation of Dryness Fraction  Actual Example, Compound Engine  Complete Entropy Diagram (Professor Boulvin's Chart).  Application of Complete Entropy Diagram  θ φ Diagram for Carnot Cycle  Condensation during Adiabatic Expansion  Adiabatic Expansion of Wet Steam  CHAPTER IV.—Heat Losses.  Effect of Steam Jacketing  Theoretical Entropy Diagram for Superheated Steam.  Effect of Superheating.	21 23 28 33 34 35 37
10. 11. 12. 13. 14. 15. 16.	Calculation of Dryness Fraction Actual Example, Compound Engine Complete Entropy Diagram (Professor Boulvin's Chart). Application of Complete Entropy Diagram θ φ Diagram for Carnot Cycle Condensation during Adiabatic Expansion Adiabatic Expansion of Wet Steam  CHAPTER IV.—Heat Losses.  Effect of Steam Jacketing Theoretical Entropy Diagram for Superheated Steam  Effect of Superheating.  Effect of Speed	21 23 28 33 34 35 37 39 46 49 56

#### CONTENTS.

		CHAPTER V.—APPLICATION TO THE GAS ENGINE.		
			P	AGE
2	4.	General Considerations		78
2	25.	Diagram for Theoretical Gas Engine		76
2	26,	Diagram for Actual Gas-engine Trial		81
		Corrected Diagram for Gas-engine Trial		80
		Constant Volume Curves		
2	19.	Heat Losses in 7 Horse Power Gas Engine		94
		CHAPTER VI.—APPLICATION TO OIL AND AIR ENGINES,		
3	80.	Diagram for 20 Horse Power Diesel Motor		97
		Stirling's Hot-air-Engine		
		Eriesson's Hot-air Engine		
		Entropy Diagram for Refrigerators		
		APPENDIX,		
1	We	ight of Dry Saturated Steam		109

#### INTRODUCTION.

The following contribution to the temperature-entropy method of graphically solving thermo-dynamic problems marks a further step in advance in the practical application of the system to the every-day questions that arise in the study of the steam engine and other heat motors. Although the method was foreshadowed by William Gibbs in 1873, it is only within the last few years that it has been applied in practice. Unfortunately, information respecting it is scattered about in the Proceedings of the Institution of Civil Engineers, in those of the Institution of Mechanical Engineers, and in various technical journals, both British and foreign. The Author has collected this information together, and has produced a work which treats the matter in a comprehensive manner, bringing it up to date so far as published materials allow.

Though not prepared to endorse every view expressed, I can fully recommend the book as likely to be eminently helpful to those studying the subject for the first time.

H. R. SANKEY, R.E.



## THE ENTROPY DIAGRAM AND ITS APPLICATIONS.

#### CHAPTER I .- ENTROPY.

#### 1. Introduction.

THE representation of various forms of energy by means of a diagram has long been known and used with advantage by engineers and others. It is usually shown as a closed figure, the area of which represents energy (in either work, heat, or other units), and the ordinates, pressure or resistance overcome, and space passed through. The ordinary indicator diagram is perhaps the most common example of such figures, but it does not show the reception and distribution of heat which takes place in all steam and other heat engines. Considering the great advance made by the science of thermo-dynamics in recent years, it is somewhat surprising to find that the representation of the thermal changes which take place in all heat engines, in the form of a "heat diagram," has been so little applied for practical use. relative advantages of the mathematical and graphical methods of representing the result of any process are so well known, that it will be unnecessary to refer to them here; beyond stating that where both methods can be employed the graphical very often becomes a useful adjunct to the mathematical, and is usually more easily grasped and understood by draughtsmen, students, and others.

The introduction of entropy diagrams is mainly due to Mr. J. Macfarlane Gray, who, in a paper read at the Paris meeting of the Institution of Mechanical Engineers in 1889,

first showed the method of representing the heat contained by water, steam, and various ideal substances, on what he termed a "theta-phi chart"; the vertical ordinates of which represented temperature, and the horizontal entropy. By denoting all absolute temperatures by the Greek letter  $\theta$  (theta), and all quantities of entropy by the letter  $\phi$  (phi), the diagram has come to be known in England as the "theta-phi diagram." In this book it will be preferable to adopt the Fahrenheit scale of temperature as that most used by practical men, and therefore absolute temperatures (Fah.) will usually be denoted by the Greek letter  $\tau$  (tau), in accordance with most of the recent works on thermodynamics.

The practical application of the entropy diagram is perhaps more due to the late Mr. P. W. Willans, who, in a paper on "Non-condensing Steam-engine Trials," read before the Institution of Civil Engineers in 1888,\* first used the diagram for the representation of steam-engine performance. Since then Capt. H. R. Sankey, R.E., has extended the subject by applying the diagram to the marine engines tested by the Research Committee of the Institution of Mechanical Engineers; the but there still seems to be a dearth of information on how to draw the diagrams that, it is hoped, the present book will, in a measure, supply.

#### 2. ENTROPY DIAGRAMS.

The theta-phi or temperature-entropy diagram is a graphic representation of the thermal changes which take place in a steam-engine cylinder during one cycle. It is plotted on a theta-phi chart, by calculations made from the mean indicator diagram of an experiment. As an aid to the thermo-dynamic study of the steam engine, it is much more useful than the better known indicator diagram, as it shows at a glance the thermal efficiency of the engine. The ordinary indicator

Sec Proc. Inst. of Civil Engineers, vol. xciti., Part III.

diagram only shows the amount of work done, independent of the amount of steam used; but the theta-phi (or  $\theta$   $\phi$  as it is better written) diagram shows, by its area, the proportion of heat utilised to the heat received.

#### Comparison of $\theta \phi$ Diagram with Indicator Diagram.

In the ordinary indicator diagram the area represents the work done on the piston in one stroke, the vertical ordinates being "pressure," and the horizontal abscisse "distance moved through." Similarly, in the  $\theta \phi$  diagram, the area

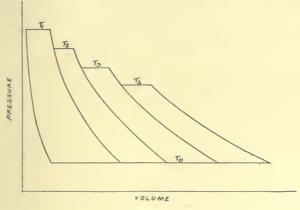


Fig. 1.

also represents work done, but in heat units; the vertical ordinates being absolute temperature (denoted by the Greek letter  $\theta$ ), and the horizontal dimension is what Clausius termed "entropy," named by Zeuner "heat weight," and represented by the Greek letter  $\phi$ . In the indicator diagram the area represents the product of pounds (pressure) × feet (distance), or foot-pounds (work); so in the  $\theta$   $\phi$  diagram the area represents the product of  $\theta$  (absolute temperature) ×  $\phi$  (entropy) in thermal units (work).

#### 3. Entropy.

Entropy is the co-ordinate with temperature of energy; *i.e.*, it is the length upon a diagram whose height is absolute temperature, and whose area is heat units.

The meaning of the term will be better understood in the light of Carnot's principle of efficiency applied to a reversible cycle in which the heat is received at various temperatures. Fig. 1 represents such a cycle, where, in the first stage, heat is received at  $\tau_1$  and discharged at  $\tau_0$ ; in the second stage, a further quantity of heat is received at  $\tau_2$  and discharged at  $\tau_0$ , and so on. Let  $Q_1$ ,  $Q_2$ ,  $Q_3$ , &c., represent the various quantities of heat received at each stage. The area of the whole figure, which represents in heat units the work done W, will equal

$$W = \frac{Q_1}{\tau_1} (\tau_1 - \tau_0) + \frac{Q_2}{\tau_2} (\tau_2 - \tau_0) + \frac{Q_3}{\tau_3} (\tau_3 - \tau_0) + \dots$$

or, for a general formula,

$$W = \frac{Q}{\tau} (\Delta \tau)$$
, in heat units;

where

Q = heat received,

 $\tau$  = temperature of reception (absolute),

 $\Delta \tau =$  difference of temperature between reception and rejection of heat.

Let R = the heat rejected to the cold body, then Q = W + R;

but 
$$Q = Q_1 + Q_2 + Q_3 + \dots$$
.

and W = 
$$\frac{Q_1}{\tau_1} (\tau_1 - \tau_0) + \frac{Q_2}{\tau_2} (\tau_2 - \tau_0) + \frac{Q_3}{\tau_3} (\tau_3 - \tau_0) +$$

or, by difference, 
$$R = \frac{Q_1 \tau_0}{\tau_1} + \frac{Q_2 \tau_0}{\tau_2} + \frac{Q_3 \tau_0}{\tau_2} + \cdots$$
.

or, 
$$\frac{R}{\tau_0} = \frac{Q_1}{\tau_1} + \frac{Q_2}{\tau_2} + \frac{Q_3}{\tau_3} + \dots . . .$$

or, 
$$\frac{Q_1}{\tau_1} + \frac{Q_2}{\tau_2} + \frac{Q_3}{\tau_3} + \dots - \frac{R}{\tau_0} = 0$$
.

or (the sum) 
$$\Sigma \frac{Q}{\tau} = O$$
.

That is to say, the algebraical sum of the changes of entropy in any complete reversible cycle is nil; therefore the entropy diagram for any reversible cycle must be a closed figure; and the final quantity of entropy will be equal to the initial, no matter where the process be started. Entropy, therefore, is the fraction or ratio  $\frac{Q}{\tau}$ , representing the amount of heat taken up or rejected by a body, divided by its absolute temperature at that time. It can be calculated from any arbitrary zero of temperature; for water and steam, it will be found most convenient to calculate the entropy above that already possessed by 1 lb. of water at 32 deg. Fah., so as to avoid including the latent heat of water. It is usually plotted with temperature as ordinates to an abscissa of entropy.

#### 4. THEORETICAL ENTROPY DIAGRAM FOR STEAM.

The thermal changes which take place when water is evaporated, expanded (as steam), and condensed, are represented on the  $\theta \phi$  diagram in the following way: Take 1 lb. of water at the condenser temperature, say 73 deg. Fah. absolute, and let A represent its position on the diagram, fig. 2, its temperature  $\tau_3$  and entropy Oa being known. Now pump it into the boiler, where it is heated from  $\tau_3$  to  $\tau_1$  deg. Fah., and, as its temperature is increased, the process will be shown by an upward curve AFB, to correspond with the vertical temperature scale; but, as it also receives heat, the curve must progress to the right to indicate its increased entropy. The change in its state by heating it from  $\tau_3$  to τ, deg. is therefore shown by the curve A FB, every increment of temperature being accompanied by a corresponding increase in entropy, denoted by  $\frac{h}{-}$ , where h is the heat contained in 1lb. of water at 7 deg. absolute temperature. Having reached the temperature  $\tau_1$  of the water in the boiler, it begins to evaporate, its temperature remains constant, but it receives an amount of heat (L1) known as

the latent heat of 1lb. of steam at  $\tau_1$  deg. temperature. This is represented in fig. 2 by the horizontal line BC—horizontal, because its temperature does not change during the operation; the amount bc of its increase in entropy

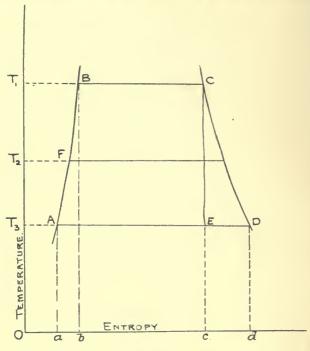


Fig. 2.

being equal to its latent heat divided by its absolute temperature, or  $\frac{L_1}{\tau_1}$ . Its total entropy Oc is made up of the two quantities Ob and bc, representing the heat of formation as water (usually denoted by h), and the latent heat L, respectively; its total heat H being the sum of the

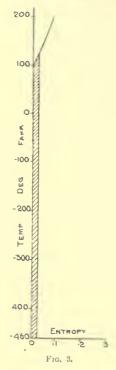
two, or h + L. At C the steam is admitted into the cylinder, and allowed to expand. If external heat be added to it during expansion, so as to keep it up to saturation point, it will follow the law of the saturation curve  $p v^{1.0625} = a$  constant, and its entropy will be denoted by the curve CD, such that all horizontal dimensions from AB to CD are equal to the latent heat of 1 lb. of dry saturated steam divided by its absolute temperature. If the steam expands adiabatically, the entropy curve will be a straight vertical line CE; because, as the steam neither receives nor loses heat, its entropy will be unchanged. This also shows clearly the amount of wetness which always accompanies adiabatic expansion, and the ratio of A E to AD represents the dryness fraction of the steam at the end of expansion. To indicate the condensing operation, the curve returns along the horizontal line from D to A, the temperature of the mixture remaining constant at  $\tau_3$  deg., and its entropy being reduced from Od to Oa.

#### CHAPTER II .- ENTROPY OF WATER AND STEAM.

#### 5. METHOD OF CONSTRUCTING THE CURVES.

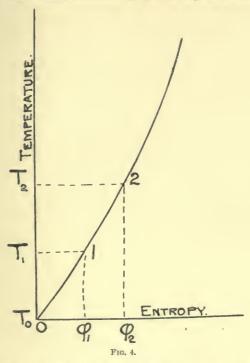
Having explained the operations to which the theoretical entropy diagram for steam refers, it is necessary to find a means of drawing the two boundary curves AB and CD of the  $\theta$   $\phi$  chart shown in fig. 2. For ordinary steam engines it is not necessary to refer to temperatures below 100 deg. Fah. (corresponding to a pressure of about 11b. absolute), nor higher than 400 deg. Fah., equal to about 260 lb. absolute pressure. We have therefore 300 deg. Fah. range of temperature to provide for, and a scale of 20 deg. Fah. to 1 in. will be found convenient if the chart be drawn on an ordinary sheet of sectional paper. For the base line, or entropy, starting from water at 100 deg. Fah., the maximum required will be 1.87; so that an entropy scale of  $0.1 \phi = 1$  in. will be ample. These scales will give for area, 1 square inch =

20 deg.  $\theta \times 0.1 \phi = 2.0$  British thermal units. It must be distinctly understood that the  $\theta \phi$  diagram is always drawn for 1 lb. of H<sub>2</sub>O, whether it be steam, water, or a mixture of both steam and water. Having plotted the scales, we start at the bottom left-hand corner of the chart with 1 lb. of



water at 100 deg. Fah., or 560 deg. Fah. absolute, and calculate the quantities of heat as above. First construct the curve of entropy of water, or aquene curve as it is sometimes called, shown by AB, in fig. 2. As this curve is almost a straight line it is only necessary to calculate the value of

entropy for some 10 or 12 points, so commencing with water heated from 100 deg. Fah. to 125 deg. Fah. (see fig. 3), the heat given to 1 lb. of water to raise its temperature from 560 deg. Fah. absolute to 585 deg. will be represented on the



 $\theta$   $\phi$  diagram by the area shaded in fig. 3. The area down to the absolute zero of temperature must be added, so as to include all the heat contained in the water. In this case, the heat received is 25.08 B.T.U. (see tables of Properties of Saturated Steam), and, therefore, the increase of entropy will be  $\frac{25.08}{572.5}$ , or 0.0438. Similarly, the value of  $\phi$  can be calcu-

lated for any temperature, and should be tabulated as in Table I., both for the purpose of plotting the aquene curve, and for future reference. Table I. gives the values of the entropy of 1 lb. of water for every 10 deg. Fah. from 32 deg. Fah. to 400 deg. Fah.

#### 6. ENTROPY OF WATER.

The aquene curve can also be plotted from calculations made by the aid of the calculus in the following manner: For any increase of temperature from  $\tau_0$  to  $\tau_1$  (see fig. 4),

$$\phi_1 = \frac{\Delta h}{\tau}$$
,

where  $\Delta h$  represents the heat necessary to raise 1lb. of water from  $\tau_0$  to  $\tau_1$ ; and  $\tau$  is the mean absolute temperature during the operation. Integrating this, we get—

$$\phi_1 = \int_{-\tau}^{\tau_1} \frac{dh}{\tau} \; ;$$

and, assuming the specific heat of water as unity,

$$dh = d\tau$$
;

or

$$\phi_1 = \int_{-\tau_1}^{\tau_1} \frac{dt}{\tau};$$

and, solving this, we get

$$\phi_1 = \log_{\ell} \tau_1 - \log_{\ell} \tau_0 ;$$

or

$$\phi_1 = \log_e \frac{\tau_1}{\tau_0};$$

that is, d  $\phi$ , or any small difference in the entropy of 1 lb. of water at any two temperatures is equal to the difference of the hyperbolic logarithms of the absolute temperatures. If the result be multiplied by the mean specific heat of water between the temperatures  $\tau_1$  and  $\tau_0$ , the formula becomes

$$\phi_{\tau_1} - \phi_{\tau_0} = s (\log_{\epsilon} \tau_1 - \log_{\epsilon} \tau_0),$$

where s represents the heat necessary to raise 1 lb. of water 1 deg. Fah., between  $\tau_0$  and  $\tau_1$ , as compared with the heat

required to raise 1 lb. of water from 39 deg. Fah. to 40 deg. Fah. The values of s are given in Table I., page 12, together with the increase of entropy for every 10 deg. Fah., calculated by the above formulæ, and the total entropy above water at 32 deg. Fah. The last column in Table I. gives the difference of entropy per 1 deg. Fah., for the purpose of interpolation.

#### 7. ENTROPY OF STEAM.

To draw the entropy curve for steam (CD, in fig. 2), an amount of entropy equal to  $\frac{L}{\tau}$  must be added to the aquene curve, where L is the latent heat of 1 lb. of steam at  $\tau$  deg. absolute temperature. The values of L and  $\frac{L}{\tau}$  are given in Table II., page 13, for every 10 deg. Fah. from 32 deg. to 400 deg., together with the difference of entropy of steam per 1 deg. Fah. for interpolating, and the total entropy of steam and water (above 32 deg. Fah.) with its difference per 1 deg. It should be noted that  $\phi_{w+s}$ , the entropy of water and steam at any temperature, is not equal to  $\frac{H}{\tau}$ , where H = total heat of evaporation from 32 deg. Fah. at absolute temperature  $\tau$ ; because h, the sensible heat of the water, is not all received at temperature  $\tau$ , but at a gradually increasing temperature.

#### 8. ENTROPY DIAGRAM FOR ICE, WATER, AND STEAM.

The entropy diagram for ice, water, and steam is shown in fig. 5, page 14. The curve AB, for ice, is drawn on the assumption that its specific heat is 0.504 at all temperatures. The heat given to 1 lb. of ice per 1 deg. Fah. rise of temperature is, therefore, equal to 0.504 B.T.U.,

or 
$$h = 0.504 d t;$$

$$\cdot \frac{h}{\tau} = \phi = 0.504 \frac{d t}{\tau};$$
or 
$$\phi_2 - \phi_1 = 0.504 \times \log_{\epsilon} \frac{\tau_2}{\tau}.$$

TABLE I.

Entropy of water, from 32 deg. Fah. to 400 deg. Fah., calculated from the formula,  $\phi_{\tau_1} - \phi_{\tau_0} = s (\log_e \tau_1 - \log_e \tau_0)$ .

Tempera- ture. Deg. Fab.	Absolute tempera- ture.	Specific heat.	Increase of entropy per 10 deg. Fah.	Total entropy above water at 32 deg. Fah.	Difference of entropy per 1 deg. Fal
t.	τ.	8.	$\phi_{\tau_1} - \phi_{\tau_0}$ .	$\phi_{vo.}$	$\Delta \phi_{w}$ .
32	492				
40	500	1.0	0.01013	0.01613	0.00202
50	510	1.0	0.01980	0.03593	0 00198
60	520	1.0	0.01942	0.05535	0*00194
70	530	1.001	0.01908	0.07441	0.00191
80	540	1.001	0.01871	0.09312	0.00187
90	550	1.002	0.01838	0.11150	0.00184
100	560	1.002	0.01806	0.12956	0.00181
110	570	1.003	0.01775	0.14731	0.00177
120	580	1.004	0.01745	0.16476	0.00174
130	590	1.004	0.01716	0.18192	0.00172
140	600	1.005	0.01689	0.19881	0.00169
150	610	1.006	0.01662	0.21543	0.00166
160	620	1.007	0.01637	0.23180	0.00164
170	630	1.008	0.01613	0.24793	0.00161
180	640	1.009	0.01589	0.26382	0.00159
190	650	1.010	0.01566	0.27948	0.00157
200	660	1:011	0.01544	0.29492	0.00154
210	670	1.012	0.01522	0.31014	0.00152
220	680	1.013	0.01501	0.32515	0.00150
230	690	1.014	0.01480	0.33992	0 00148
240	700	1.015	0.01461	0.35456	0.00146
250	710	1 017	0.01443	0.36899	0.00144
260	720	1.019	0.01425	0.38324	0.00142
270	730	1.021	0.01408	0.34732	0.00141
280	740	1.022	0.01391	0.41123	0.00139
290	750	1.024	0.01374	0.42497	0.00137
300	760	1.026	0.01358	0.43855	0 00136
310	770	1.027	0.01343	0.45198	0.00134
320	780	1.029	0.01328	0.46526	0.00133
330	790	1.031	0.01313	0.47839	0.00131
340	800	1.032	0.01298	0.49137	0.00130
350	810	1.034	0.01284	0.50421	0.00128
360	820	1.036	0.01271	0.51692	0.00127
370	830	1.038	0.01258	0:52950	0.00126
350	840	1.040	0.01245	0.54195	0 00124
3:10	850	1.042	0.01233	0.55428	0.00123
400	860	1.044	0.01221	0.56649	0.00122

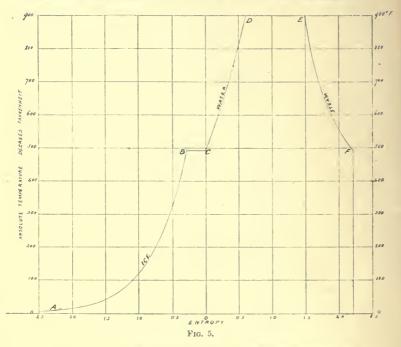
TABLE II.

Entropy of Dry Saturated Steam, from 32 deg. Fah.
to 400 deg. Fah.

Tempera-	Absolute tempera-	Latent heat.	Entropy of 11b.	Difference of entropy		tropy of steam.
Deg. Fah.	ture.	neat.	steam.	per 1 deg. Fah.	Per lb.	Difference per 1 deg. Fah.
<i>t.</i>	τ.	L.	$\phi_{\mathfrak{s}} = \frac{\mathrm{L}}{\tau}.$	$\Delta \phi_{s}$ .	$\phi_{w.} + \phi_{s.}$	$\Delta \phi_{10} + s.$
32	492	1091-7	2 2189	0°005S1	2-2189	0.00380
40	500	1086-2	2.1724	0.00561	2.1885	0.00363
50	510	1079 3	2.1163	0.00542	2.1522	0.00348
60	520	1072.3	2.0621	0.00521	2:1174	0.00330
70	530	1065.3	2.0100	0.00502	2.0844	0.00312
80	540	1058 3	1.9598	0.00483	2.0529	0.00299
90	550	1051.3	1.9115	0.00465	2.0230	0 00285
100	560	1044.36	1.8649	0.00449	1:9945	0.00272
110	570	1037:30	1.8200	0.00434	1.9673	0.00259
120	580	1030.42	1.7766	0.00420	1.9414	0.00249
130	590	1023.40	1.7346	0.00406	1.0165	0.00237
140	600	1016:39	1.6940	0.00393	1.8928	0.00227
150	610	1009:38	1:6547	0 00379	1.8701	0.00216
160 170	620 630	1002:37 995:33	1.6167	0.00368	1.8485	0.00207
180	640	988:30	1:5799 1:5442	0.00357	1 *8278 1 *8080	0.00198
190	650	981-24	1:5096	0.00346	1.7891	0.00189
200	660	974.18	1.4760	0.00336	1.7709	0 00182
210	670	967-10	1.4434	0.00326	1.7535	0.00174
220	680	960.03	1.4118	0.00316	1.7369	0.00166
230	• 690	952-94	1.3811	0.00307	1 7211	0.00158
240	700	945.83	1.3512	0.00299	1.7058	0.00153
250	710	938.76	1.3220	0.00295	1.6910	0.00148
260	720	931:56	1.2938	0.00282	1.6770	0.00140
270	730	924.41	1.2663	0.00275	1.6636	0.00134
280	740	917:25	1.2395	0.00268	1.6507	0.00129
290	750	910.06	1.2134	0.00261	1.6384	0.00123
300	760	902.86	1.1880	0.00254	1.6265	0.00110
310	770	895.64	1.1632	0 00248 0 00242	1 6152	0 00113
320	780	888:41	1.1390	0.00242	1.6043	0.00100
330	790	881.15	1.1124	0.00236	1.5938	0.00102
340	800	873.88	1.0923	0 00225	1.5837	
350	810	866.58	1.0698	0.00219	1.5740	0.00097
360	S20	859-27	1.0479	0.00215	1.5648	0.00080
370	830	851.95	1.0264	0.00210	1:5559	0.00085
380	840	844.58	1:0054	0.00205	1.5474	0.00083
890	850	837:20	0.9849	0.00200	1.5392	0.00078
400	860	829:84	0.9649	0 00=00	1.5314	0 00013

Thus  $\phi$  becomes infinity when  $\tau = 0$ ; and the real origin of the entropy scale should be at infinity on the left-hand side of the figure. For convenience of plotting the origin

has been taken as for water at 492 deg. Fah. (absolute), the amount of entropy on either side of this point being considered as positive and negative quantities. The break in the curve between B and C is due to the latent heat of water, or the heat required by 1 lb. of ice at 492 deg. for



conversion into water at 492 deg. Its amount is equal to 143 B.T.U., and its entropy is therefore equal to  $\frac{143}{492} = 0.29$ .

The curves for water and steam are drawn as already explained. If the two curves CD and FE be produced, they will meet at a point known as the "critical temperature" for water and steam, above which there can be no

liquid. Mr. Macfarlane Gray\* finds this to be at about 750 deg. Cen., or about 1,350 deg. Fah. (absolute). The area of the diagram, fig. 5, is such that each square represents  $100 \times 0.5 = 50$  B.T.U.

#### 9. CONSTANT VOLUME CURVES.

Having constructed the two boundary curves of the  $\theta \phi$  chart (the most tedious and difficult part of the process), it will be advisable to draw "constant pressure lines" and "con-

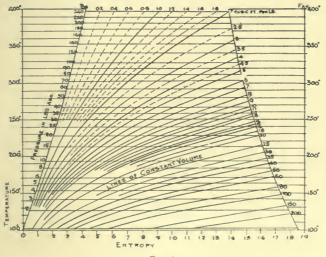


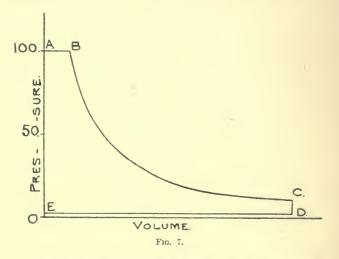
FIG. 6.

stant volume curves," as shown in fig. 6, in order to facilitate the transfer of indicator diagrams. The former are simply horizontal lines drawn across the chart between the two boundary curves, at a height corresponding to the temperature of dry saturated steam, as given in the steam tables. If a temperature scale of 20 deg. Fah. to 1 in. be adopted, the

<sup>&</sup>quot; See Proc. Inst. of Mechanical Engineers, July, 1889, page 419.

pressure lines may be drawn for every 1 lb. pressure up to about 20 lb. absolute, but beyond this they should be reduced to every 2 lb., and afterwards to 5 lb. intervals, to prevent complicating the chart. These constant pressure lines are shown in fig. 6.

The constant volume curves present a little more difficulty. They represent the loss of entropy due to a drop in pressure at constant volume, or that which takes place in the ordinary engine cylinder at release, when it opens to the condenser.



For example, suppose 1 lb. of steam at 100 lb. pressure to expand, keeping dry, to 10 lb. absolute, and then to open to the condenser so that the pressure falls to 2 lb. absolute at constant volume, as shown in fig. 7. The 1 lb. of dry steam at 10 lb. pressure occupies 37.8 cubic feet, and has an entropy of 1.64 above water at 100 deg. Fah.; its pressure falls to 2 lb. by condensing, without any increase in its volume, thus causing a reduction of entropy in the steam. At 2 lb. pressure it would occupy 173 cubic feet if it were

all steam; but we know that it only occupies 37.8 cubic feet, and therefore there can only be  $\frac{37.8}{173}$ , or 0.218 lb. of steam in the cylinder after opening to the condenser, which will

in the cylinder after opening to the condenser, which will possess an entropy represented by the distance E D, in fig. 8, where

$$\frac{\mathrm{ED}}{\mathrm{EF}} = 0.218 \; ;$$

the curve CD forming a part of one of the constant volume curved lines which have to be drawn. To construct these

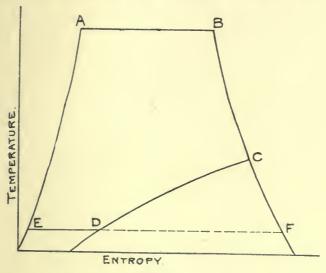


Fig. S.

curves, divide the horizontal distance or entropy between the water and steam curves, at any particular pressure, into as many equal parts as there are cubic feet of volume to 1 lb. of dry saturated steam at that pressure. For example, at 60 lb. absolute pressure the specific volume of steam is practically 7 cubic feet; and therefore the curves for 1, 2, 3, 4, 5, and 6 cubic feet constant volume lines (fig. 6) all cut the 60 lb. pressure line at equal horizontal distances from one another.

Another method of drawing the constant volume curves is based on the principle that

$$\frac{d P}{d \tau}$$
 varies as  $\frac{L}{\tau}$ , or varies as  $\phi$ .

This is found by equating the work done to the heat expended in a perfect steam engine working on the Carnot

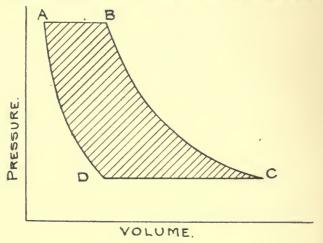


Fig. 9.

cycle. Take 1 lb. of water of volume O A, fig. 9, and pressure  $P_1$ ; evaporate it to dry steam of volume O B and temperature  $\tau_1$ ; expand it adiabatically to C, fig. 9, where its pressure is  $P_2$ , and temperature  $\tau_2$ ; now condense it at this pressure and temperature until it occupies the volume O D; and, finally, complete the cycle by compressing it adiabatically until it occupies its initial volume of O A.

The efficiency of the cycle will be  $\frac{\tau_1 - \tau_2}{\tau_1}$ ; and the work done

will be equal to the area shaded in fig. 9,

$$= \mathbb{L}\left(\frac{\tau_1 - \tau_2}{\tau_1}\right)$$

in heat units, or J. L  $\left(\frac{\tau_1 - \tau_2}{\tau_1}\right)$  in work units. Now, if we make  $P_1 - P_2 = d$  P very small, and  $\tau_1 - \tau_2 = d$   $\tau$  also very small, the work done will be equal to the rectangle A B C D, which equals

(vol. O B - vol. O A) . 
$$(P_1 - P_2)$$
  
=  $(V - v) d P$ ,

where V and v are the volumes of 1 lb. of steam and 1 lb. of water respectively.

But the work done is equal to the heat expended; therefore

$$(V - v) \cdot dP = J \cdot L\left(\frac{d\tau}{\tau_1}\right)$$

Transposing J, and dividing all through by  $d\tau$ , we get—

$$\frac{\mathbf{V} - \mathbf{v}}{\mathbf{J}} \times \frac{d\mathbf{P}}{d\tau} = \frac{\mathbf{L}}{\tau} = \phi.$$

But v = approximately 0.017 cubic feet, and V is a constant for any pressure, and J a numerical constant; therefore, for any particular pressure,

$$\frac{d P}{d \tau}$$
 varies as  $\frac{L}{\tau}$ , or varies as  $\phi$ .

In other words, the horizontal ordinates of any constant volume curve at various pressures are proportional to the value of  $\frac{d}{d\tau}$  for those particular pressures. For example,

take 1 lb. dry steam at 150 lb. absolute pressure, which has a specific volume of 2 978 cubic feet. The value of

$$\frac{d P}{d \tau} = 1.91$$

(see Table IA., Appendix of Cotterill on the "Steam Engine"), and  $\stackrel{L}{=}$  or  $\phi$  for steam alone (see Table II., page 13)

$$=\frac{860.62}{818.16}=1.052.$$

At 100 lb. pressure,

$$\frac{d P}{d \tau} = 1.39,$$

and therefore the  $\phi$  ordinate for the 2.978 constant volume curve will be

$$\frac{1.39}{1.91} \times 1.052 = 0.765.$$

The values of  $\phi$  for 2.978 cubic feet of steam calculated in this way are given in Table III., and if they are plotted on the  $\theta$   $\phi$  chart at their proper pressures, they will be found to form a constant volume curve for 2.978 cubic feet, which will be just inside the 3 cubic feet curve already drawn by the geometrical method previously explained.

TABLE III.

Value of Entropy for 2.978 cubic feet of steam, for drawing constant volume line.

Absolute pressure. Pounds per square inch.	$\begin{array}{c} \textbf{Temperature} \\ \textbf{Fahrenheit.} \\ \textbf{Degrees.} \\ t \end{array}$	Ratio of $\frac{d p}{d t}$ from steam tables.	Entropy of 2.978 cubic feet of steam by proportion.
150	358:16	1.91	1.05
100	327:57	1:39	0.765
50	280.85	0.79	0.435
20	227.92	0.376	0.207
10	193.24	0.213	0.117
5	162:33	0.119	0.065
2	126:27	0.055	0.030

#### CHAPTER III.—Conversion of Indicator Diagram to Entropy Diagram.

#### 10. CALCULATION OF DRYNESS FRACTION.

To convert the ordinary indicator diagram to the  $\theta \phi$  diagram, it will be necessary to have the following data furnished by an experiment:—

(a) Exact size of cylinders, and clearance volumes of each

end of each cylinder.

(b) Dryness fraction of the steam in each cylinder at any one period during expansion.

(c) A mean indicator diagram for each cylinder.

The diameters of the cylinders should be taken from gauges, and corrected to allow for the expansion of the cylinders when hot. The clearance volumes can be obtained either by calculation or direct measurement by filling the ports, &c., with water. When the latter method is employed, care must be taken to allow for the escape of the air; and where possible, it is advisable to calculate the volume as well as measure it, the one method serving as a check upon the other.

The dryness fraction is calculated from the mean indicator diagram, when the weight of steam used by the engine per stroke is known, in the following manner: Take any point in the expansion period of the stroke, as A, fig. 10, and scale the pressure  $P_a$  shown by the indicator diagrams at that point. Calculate the volume  $(V_a)$  occupied by the steam in the cylinder at that portion of the stroke; i.e., if A be taken at half stroke,

 $V_a = \frac{A \times L}{2} + v,$ 

where A = mean area of front and back of piston in square feet;

L = length of stroke in feet;

v = mean clearance volume of front and back in cubic feet. Multiply the volume  $V_a$  by the weight of 1 cubic foot of steam at  $P_a$  pounds absolute pressure, and obtain the weight  $(W_a)$  of steam per stroke shown to be present in the cylinder at point A. Subtract from this weight  $(W_a)$ , the weight of steam  $(W_b)$  left in the cylinder from the previous stroke at the end of compression, as shown at B, fig. 10. This will be equal to  $V_b \times$  density of steam at  $P_b$  pounds absolute pressure; and the *net* weight of steam per stroke accounted for by the indicator at point A will be  $W_a - W_b$  pounds. Compare this with the net weight of steam used by the engine per stroke, which can be obtained by measuring

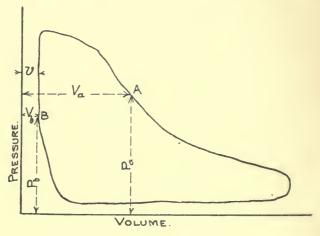


Fig. 10.

the feed water and allowing for steam pipe and jacket condensation, or by measuring the water discharged by the air pump if a surface condenser be used, and the proportion

of indicated steam is the required dryness fraction at

point A. A table of the weight of dry saturated steam for various pressures will be found in the appendix.

To obtain an average indicator diagram for each cylinder, all the diagrams taken during the trial should be divided into 20 parts, and the forward and backward pressures marked off on long strips of paper, the means so obtained being tabulated as shown in Table V., page 25, together with the volume of the cylinder plus clearance at that part of the stroke.

#### 11. ACTUAL EXAMPLE: COMPOUND ENGINE AT HAMPTON.

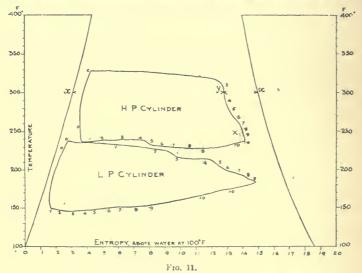
As an example, take the trial of a compound pumping engine at Hampton, of the Southwark and Vauxhall Waterworks, and tested by Prof. Hudson Beare in January, 1894. Full particulars of the trial will be found in the third report of the Steam Jacket Research Committee of the Institution of Mechanical Engineers, published in October, 1894. The engine is of the vertical, inverted, surface-condensing type,

TABLE IV.—TRIAL OF COMPOUND ENGINE AT HAMPTON.

Data from Proc.Inst. Mech. E., October, 1994.

	H.P. cylinder.	L.P. cylinder
Total volume of cylindercubic feet	38*41	105.05
Volume of cylinder at releasecubic feet	36.49	99-80
Volume of clearance	0.86	1.94
Total volume of steam at releasecubic feet	37.35	101:74
Pressure at release	26-12	8.02
Density of steam at above pressure	0.06515	0.02145
Weight of steam present at release W lbs.	2.4332	2.1822
Pressure at end of compressionlbs. absolute $p_2$	80.1	12:42
Density of steam at above pressure	0.18645	0.03239
Weight of steam left in clearance	0.1600	0.0628
Net steam accounted for by indicator( $W - ic$ ) lbs.	2-2723	2.1194
Dryness fraction at releaseper cent	92.1	85 9

with cylinders 32 in. and 52\frac{5}{8} in. diameter; stroke, 7 ft.; piston rods, 6 in. diameter. Considering only the trial with all the steam jackets on, made on January 4th, 1894, Table LX., page 578 of the report, gives the particulars from which the figures given in Table IV. are taken. From the mean indicator diagrams published in plate 143 of the committee's report, the pressures at every one-twentieth part of the stroke in both cylinders have been scaled, and tabulated in



Tables V. and VI. To calculate the  $\theta$   $\phi$  volume—that is, the volume which 1 lb. of steam of the required dryness would occupy—any initial point may be taken where the dryness fraction of the steam is known; as at release, or 95 per cent of the forward stroke. In this particular trial, the weight of steam used was 6,284 lb. per hour, or 2:4665 lb. per stroke of steam and water passing through the cylinders, as found by measuring the air-pump discharge; and at release, in the high-pressure cylinder, this weight of steam occupies 37:35

TABLE V. $-\theta \phi$  Diagram for Compound Engine at Hampton. High-pressure Cylinder,

Portion of	Pressure in pounds absolute.		Actual volume.	$\theta \phi$ volume	
stroke.	Forward stroke.	Backward stroke.	Cubic feet.	(for 1 lb. steam) Cubic feet.	
0	99 2	80.1	0.86	0.32	
.05	98.3	32.0	2 78	1.02	
-1	96.6	22.8	4.70	1.78	
•15	93.5	22.8	6.62	2 50	
•2	89.7	23.0	8:54	3*23	
•25	84.2	23.4	10.46	3.96	
•3	74.8	23.6	12:38	4*691	
•35	63.5	23.8	14.30	5.41	
-4	57.0	23.8	16 22	6.14	
.45	51.0	23*4	18.14	6.87	
•5	46.1	22.7	20.06	7.59	
*55	41.9	21.9	21.98	8.32	
-6	38-6	21.2	23-90	9.05	
*65	35*8	20.3	25.82	9.77	
•7	33.8	19.8	27:75	10.20	
:75	31.0	19:4	29.67	11.23	
.8	29.8	19.2	31.59	11.96	
*85	28.4	19-1	33.51	12.69	
•9	27:2	19:4	35*48	13.41	
•95	26.1	19.8	37:35	14.14	
1.0	24 0	22.5	39:27	14.87	

cubic feet, the pressure at that point being 26.12 lb. absolute. The specific volume of dry steam at this pressure is 15.35 cubic feet per pound; but knowing that the steam is not dry at this point, but possesses a dryness fraction of 0.921, and

therefore the volume of 1 lb. of steam of dryness fraction 0.921, and pressure 26.12 lb., will be

 $15.35 \times 0.921 = 14.14$  cubic feet.

TABLE VI. $-\theta \phi$  Diagram for Compound Engine at Hampton. Low-pressure Cylinder,

Portion of	Pressure in pounds absolute.		Actual volume	$\theta \phi$ volume	
stroke.	Forward stroke.	Backward stroke.	Cubic feet.	volume (for 1 lb. steam). Cubic feet.	
0	23.3	19:5	1:94	0.76	
<b>.</b> 05	23.1	4:7	7:19	2.83	
1	22.2	3.5	12.44	4.90	
<b>.</b> 15	20.8	3 5	17:69	6.96	
*2	19*5	3.5	22.95	9.03	
•25	18.1	3.5	28.20	11;10	
.3	16 7	3.5	33.45	13.17	
*35	15.6	2.2	38.70	15.23	
•4	14.8	3.6	43.96	17.30	
*45	14.0	3.6	49.21	19.37	
•5	13.3	3.6	54*46	21.44	
*55	12.7	3.6	59.72	23.51	
•6	12.0	3.6	64.97	25.57	
*65	11.2	3.7	70.22	27.64	
•7	10.5	3.7	75.47	29*71	
.75	9.8	3:7	80.73	31.78	
•8	9.2	3.8	\$5.98	33.84	
*85	8*7	3*8	91.23	35*91	
. *()	8.3	3.0	96*48	37.98	
195	8.0	4.1	101:74	40.05	
1.0	6.5	5.5	106:99	42.11	

This will be the  $\theta \phi$  volume for that particular point, and is the specific volume of steam at the pressure shown, reduced according to its dryness. The volume occupied by the 0 079 lb. of water will be 0.079  $\times$  0.017 cubic feet, and is quite negligible for all practical purposes. Knowing the  $\theta \phi$  volume at any one point, the  $\theta \phi$  volumes for the other parts of the stroke are calculated by proportion—i.e., at 10 per cent of the stroke the actual volume is 4.70 cubic feet, and the  $\theta \phi$  volume will therefore be

$$\frac{4.70 \times 14.14}{37.35} = 1.78$$
 cubic feet.

The pressures and volumes for the low-pressure cylinder are similarly dealt with, and the figures are given in Table VI. The  $\theta \phi$  diagrams for this trial are given in fig. 11.

It will be seen that in the example quoted no notice has been taken of the increase or reduction in the volume of steam due to the opening and closing of the passage in the main slide valve. This has not been possible in the present case, because the report does not mention when the main valve cut off; but if the position of this had been known, a "kick" would appear on the  $\theta$   $\phi$  diagram, somewhat as shown dotted at X, in fig. 11, due to the closing of the passage in the main valve at the end of the forward stroke. The ordinary indicator diagram does not show this.

As a check upon the accuracy with which the  $\theta$   $\phi$  diagram has been drawn, its area should be measured by a planimeter, and reduced to heat units according to the scales of temperature and entropy adopted. If the temperature scale be 20 deg. Fah. to 1 in., and entropy  $0.1 \phi = 1$  in, then the area of the  $\theta \phi$  diagram in square inches, multiplied by  $20 \times 0.1 = 2.0$ , will give the work done per stroke expressed in B.T.U. This multiplied by 772, and by the number of strokes made by the engine per minute, and divided by 33,000, should give the same I.H.P. as that calculated from the indicator diagrams for the same cylinder, in the usual way.

#### 12. COMPLETE ENTROPY DIAGRAM.

Professor Boulvin, of Gand, has investigated and published (see Engineering, January 3rd, 1896; and Revue de Mécanique, June, 1897) a complete entropy chart for steam, from which, once the curves are correctly drawn, all the remaining points can be obtained geometrically instead of by calculation. The chart is rather complicated, as it includes four separate charts, all drawn from a common centre O (see fig. 12), but it is comprehensive, and saves the continual reference to steam tables.

Let O in the centre of the diagram be the initial point. Plot a temperature scale vertically upwards on the line O W, starting with any convenient minimum temperature, such as 100 deg. Fah. or 32 deg. Fah., and adopting any convenient scale. It will be found advisable to plot the chart on a good-sized sheet of  $r^{1}v^{-}$  in. squared paper. Draw a scale of entropy O X, horizontally to the right to any convenient scale, the entropy given being above that contained in 1 lb. of water at a temperature equal to the zero adopted in the temperature scale. Volumes are plotted vertically downward on the axis O Y, to a scale to be afterward determined; and pressures (absolute) are measured horizontally to the left, as shown by the line O Z, to any convenient scale.

Starting with the first quadrant (using the word in its trigonometrical sense), the curve OAB, representing the relation between temperature and pressure of dry saturated steam, can be plotted direct from steam tables, using the temperature and pressure scales already decided upon. The entropy curves OCD and EFG, for water and steam respectively, shown in the second quadrant, are drawn as previously described in Chapter II.

To determine the volume scale O Y, take any point on the temperature pressure curve, say A at 250 deg. Fah., and draw a tangent to the curve at A, making an angle a with the vertical axis O W. From C and F in the second quadrant, representing the entropy of 1 lb. of water and

1 lb. dry steam respectively at 250 deg. Fah., draw the vertical ordinates CH and FJ on to the axis OX, and from H draw HL parallel to the tangent at A in the first

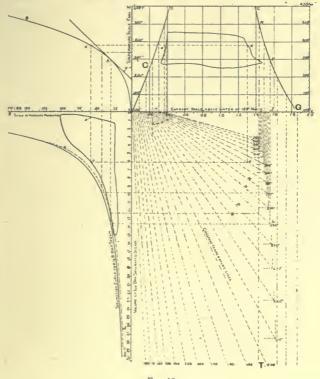


Fig. 12.

quadrant, to meet F J produced in L. Drop similar perpendiculars from other temperatures, such as shown at 200 deg., 300 deg., &c., and draw the inclined lines M N P Q, &c., parallel to tangents to the pressure temperature curve O A B at their respective temperatures.

These lines in the third quadrant represent the increasing volume of 1 lb. steam in being generated at constant temperature, and are therefore called for brevity "constant temperature lines," in contradistinction to the "constant volume lines," drawn on the  $\theta \phi$  chart by Captain Sankey's process. They should be drawn to start with a volume of 0.016 cubic feet, and not from the zero of volume scale, as shown; but the volume of 1 lb. water is so small in comparison to the volume of the steam, that it is impossible to show it on the diagram, and is quite negligible. The vertical distances J L, &c., represent to scale the volumes occupied by 1 lb. of dry saturated steam at the temperatures indicated, and furnish the necessary data for constructing the volume scale O Y.

An alternative method of drawing the constant temperature lines, involving a little elementary trigonometry, gives more accurate results than the geometrical method just described. The angle  $\alpha$  made by H L to the vertical axis O Y, is equal to the angle  $\alpha$ , made by the tangent at A to the vertical axis O W in the first quadrant. But the tangents of all such angles as these are proportional to the ratios of  $\frac{d}{d} \frac{p}{t}$ , representing increase in pressure per unit in-

crease of temperature. This ratio of  $\frac{d n}{d t}$  for steam is given

in "Cotterill on the Steam Engine" (see Appendix, Table IA) for every degree of temperature from 93 deg. Fah. to 432 deg. Fah.; but the ratio given there will only be equal to the tan a when the pressure and temperature scales are equal. If the pressure scale be drawn twice as large as the temperature scale (i.e., 1 lb. pressure = 2 deg. Fah.), as it is in fig. 12, the ratio of  $\frac{dp}{dt}$  must be doubled, in order to equal

the required tangent. Knowing the values of tan  $\alpha$  for various temperatures, the lines HL, M, N, P, Q, &c., can easily be drawn with the aid of a protractor and table of natural tangents. Table 7 gives the value of these angles

TABLE VII.—Constant Temperature Lines for Complete Entropy Diagram for Steam.

Temperature Fab., t.	$\frac{d p}{a t}$ , from "Cotterill."	Tan a for the particular scales used.	Corresponding value of a.	
100	0.02835	0.0567	Deg. min. 3 15	
110	0.03675	0.0785	4 12	
120	0.04695	0.0939	5 22	
130	0.05945	0.1189	6 47	
140	0.07430	0.1486	8 27	
150	0.09185	0.1837	10 25	
160	0.11335	0.2267	12 46	
170	0.1375	0.275	15 23	
_180	0.1665	0.333	18 25	
190	0.2005	0.401	21 51	
200	0.2385	0.477	25 80	
210	0.2825	0.565	29 28	
220	0.3325	0.665	33 37	
230	0.391	0.782	38 1	
240	0 455	0.010	42 18	
250	0.523	1.047	46 19	
260	0.603	1 -206	50 20	
270	0.691	1 383	54 8	
280	0.785	1.570	57 30	
290	0.892	1.785	60 44	
300	1.012	2.024	63 42	
310	1.135	2.27	66 14	
320	1.275	2.55	68 35	
330	1.42	2.84	70 36	
340	1.28	3.16	72 26	
350	1.75	3.50	74 3	

for every 10 deg. Fah., from 100 deg. Fah. to 350 deg. Fah. but must only be used when the pressure scale is twice as open as the temperature scale.

Knowing the volume occupied by 1 lb. of dry saturated steam at a temperature of 250 deg. Fah. (13:53 cubic feet), represented by J L, the volume scale O Y can be drawn, and the saturation curve for 1 lb. of dry steam plotted in the fourth quadrant by the aid of the pressure scale O Z. This completes the construction of the chart as used by Professor Boulvin, which "represents all the transformations of a body by the simultaneous variation of two co-ordinates, which are sufficient to characterise its state, viz., its temperature and its entropy." (See La Revue de Mécanique, June, 1897.)

It is instructive to note that the "constant temperature lines" afford a convenient method of drawing the adiabatic expansion curve for steam on the p. v. diagram, without the aid of any calculations whatever. Take 1 lb. of steam, dry, at any particular temperature—say 350 deg. Fah.: its adiabatic expansion can be shown in the fourth quadrant of fig. 12 by the following graphical method: From R, on the steam entropy curve at 350 deg. Fah., draw RS at right angles to OX, and produce it into the third quadrant to T, as shown. Where ST intersects the constant temperature lines will represent the volume occupied by the dry-steam portion of the mixture at various stages of its expansion, and can, therefore, be transferred directly to the pressure-volume curve in the fourth quadrant. By transferring the temperature into the first quadrant, and where it intersects the curve OAB, drawing ordinates into the fourth quadrant gives various points of intersection with the volume lines already drawn, thus giving the adiabatic expansion curve on the p. v. diagram for steam initially dry at 350 deg. Fah. The curve is shown dotted in fig. 12, page 29, and marked a, b, c, d.

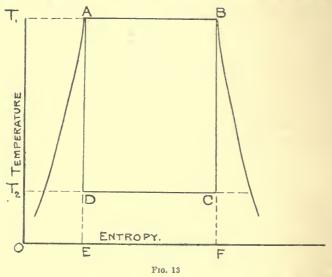
#### 13. APPLICATION OF COMPLETE ENTROPY DIAGRAM.

The practical application of the complete entropy diagram will best be explained by taking an actual steam engine test as example. Referring to the trial of a compound engine at Hampton Waterworks, by Professor Beare, already described (see page 23), the mean indicator diagram must first be plotted in the third quadrant to the scales adopted, lengthening or shortening the volume scale as required to correspond with 1 lb. of mixture present in the cylinder. In the trial under consideration, the weight of steam and water passing through the cylinders was 2:4665 lb. per stroke, as found by measuring the water discharged from the air pump. Add to this the weight of steam left in the high-pressure cylinder after compression, from the previous stroke, which is calculated to be 0.1609 lb. per stroke, and the sum, or 2.6274 lb. per stroke, is the weight of steam and water present in the high-pressure cylinder during expansion. This figure also represents a kind of volume factor by which the actual volumes must be divided, in order to obtain the volume of 1 lb. of mixture of similar dryness, for use in plotting on the  $\theta \phi$  chart. The actual volume of the high-pressure cylinder at release is 37:35 cubic feet, which, divided by 2:6274, gives 14:215 cubic feet volume at release for an imaginary cylinder containing 1 lb. of mixture of proportionate dryness. Dividing the mean indicator diagram up into any number of equal parts, say 20, of the stroke, the volumes at the other points can be readily transferred to the chart by proportional compasses, or by dividing the actual volume of the cylinder and clearance at any part of the stroke by 2.6274. The diagram being plotted, take any point e on it, and transfer into the first quadrant to meet the curve O A B, fig. 12, and thence into the second quadrant, as shown by dotted lines. Also transfer the point e into the third quadrant, and where it intersects its proper "constant temperature line" transfer into the second quadrant its intersection with the line previously

drawn at  $e^1$ , giving a corresponding point on the entropy diagram. The process of transferring the indicator diagram from the fourth to the second quadrants is one of simple projection, and in reality it takes less time to do it than is occupied in reading how to do it. It should be noted that the difficult curves to be drawn are the fixed ones, whereas the new curves required separately for each experiment are simple ones, and easily drawn with a little practice.

#### 14. θφ DIAGRAM FOR CARNOT CYCLE.

Having shown how to draw the  $\theta \phi$  diagram for a steamengine test, it will be interesting to examine the great facilities afforded by it for explaining some of the various



phenomena that occur in the steam-engine cycle. In the first place, it should be noted that the  $\theta \phi$  diagram proves graphically the efficiency of a perfect reversible cycle. Take the well-known Carnot cycle, shown on the pv or

ordinary indicator diagram in fig. 9, page 18, and on the  $\theta \phi$ diagram in fig. 13. The work done per stroke is represented in fig. 9 by the area of the figure ABCD, but neither the heat received and rejected, nor its efficiency, are shown on the pv diagram. Transfer the same series of changes to the  $\theta \phi$  diagram, shown in fig. 13. The "state point" A represents 1 lb. of water, of temperature  $\tau_1$ , and entropy O E, evaporated into 1 lb. of dry steam, of the same temperature  $\tau_1$ , but with an increase of entropy shown by EF. At B, it expands adiabatically to C, its temperature falls to To, but the amount of heat it contains is neither increased nor diminished, and therefore its entropy is the same at C as it is at B; or, the expansion is shown on the  $\theta \phi$  diagram by the vertical line BC, representing the fall in temperature from  $\tau_1$  to  $\tau_2$ . At C it is compressed isothermally, at the same temperature  $\tau_2$ , but giving up a quantity of heat denoted by its reduced entropy from OF to OE. From D to A, adiabatic compression is represented by the vertical line DA, at constant entropy OE, but with an increasing temperature from  $\tau_2$  to  $\tau_1$ . The work done during the cycle is represented in heat units by the area of the rectangle ABCD, fig. 13, and the heat received by the area of the rectangle ABFE; thus the efficiency,

or 
$$\frac{\text{work done}}{\text{heat received}} = \frac{ABCD}{ABFE} = \frac{\tau_1 - \tau_2}{\tau_1}$$
.

The line OEF corresponds to the absolute zero of temperature; and should be much lower down than indicated on the diagram. The efficiency of this cycle can be proved mathematically, but it is seen much more clearly on the  $\theta \phi$  diagram.

## 15. CONDENSATION DURING ADIABATIC EXPANSION.

The  $\theta \phi$  diagram also shows graphically the varying dryness of the steam during the expansion period of the stroke. Neglecting the volume of the water present, which (except in the case of very wet steam) is comparatively nil, the

proportion  $\frac{xy}{xx}$ , shown in fig. 11, page 24, represents the dryness fraction of the steam at y, and, similarly, the dryness at any time during expansion can be scaled off the  $\theta \phi$  diagram.

It also shows the amount of condensation due to the adiabatic expansion of steam. For instance, 1 lb. of dry steam at 170 lb. absolute pressure, expanded adiabatically to 20 lb.

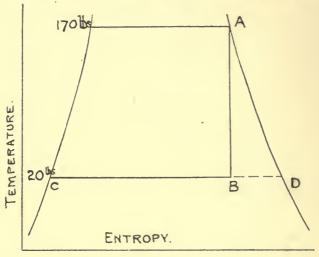


Fig. 14.

absolute, as shown by the line AB in fig. 14, will have a dryness fraction of  $\frac{C}{C}\frac{B}{D}$ , or 0.8797. But the line CB represents the latent heat of 0.8797 lb. steam at 20 lb. pressure, which should occupy a volume of

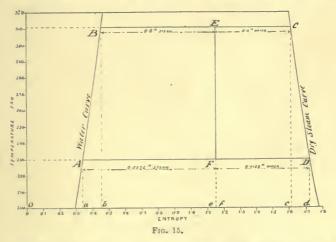
 $0.8797 \times 19.72 = 17.35$  cubic feet.

According to the law of adiabatic expansion—viz.,  $p \cdot v^{1\cdot 135} =$  constant—the volume occupied by 1 lb. of steam at 170 lb. pressure, initially dry, and expanded to 20 lb., is 17·45 cubic

feet. This difference may be due to the volume occupied by, and the heat contained in, the 0·1203 lb. water present, which is not shown by the  $\theta \phi$  diagram, but which should be allowed for when dealing with wetter steam.

#### 16. ADIABATIC EXPANSION OF WET STEAM.

If a mixture of steam and water be expanded adiabatically, it is shown on the entropy diagram by dividing the higher temperature entropy into two parts in the ratio of the dryness of the steam at the commencement of expansion. For example, take question No. 41, given in the Science and Art



Department's Examination in Steam, 1898, where 1 lb. of stuff of 0.6 dryness is expanded adiabatically from 311 deg. Fah. to 230 deg. Fah., and it is required to find the weight of water present at the end of expansion, the entropy of 1 lb. of water being given as 0.339 and 0.451, and that of 1 lb. of dry steam as 1.716 and 1.612 at the lower and higher temperatures respectively. Draw the approximate water and dry steam entropy curves AB and CD from the data given, as shown in fig. 15, and divide BC into two parts

representing to scale the dryness of the steam (0.6) at 311 deg. Fah.; that is, make

$$\frac{BE}{BC} = 0.6,$$

and drop the perpendicular EF meeting AD in F at the lower temperature. Then  $\frac{AF}{AD}$  will represent the weight of

steam, or  $\frac{F}{A}\frac{D}{D}$  the weight of water present at the end of expansion.

Using italic letters a, b, c, &c., to represent the projection of their corresponding state points on the zero temperature line, the result may be expressed numerically thus:

$$b c = O c - O b,$$

$$= 1.612 - 0.451 = 1.161;$$

$$b f = (b c) \times 0.6 = 1.161 \times 0.6 = 0.6966;$$

$$O f = O b + b f,$$

$$= 0.451 + 0.6966 = 1.1476;$$

$$a d = O d - O a,$$

$$= 1.716 - 0.339 = 1.377;$$

$$f d = O d - O f,$$

$$= O d - O e,$$

$$= 1.716 - 1.1476 = 0.5684;$$

$$\frac{f}{a d} = \frac{0.5684}{1.377} = 0.4128.$$

Answer = 0.4128 lb. water at end of expansion.

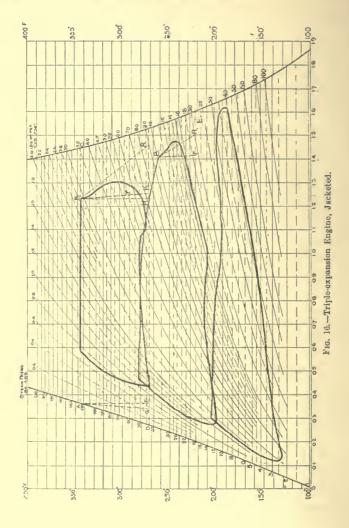
It is clearly seen from fig. 15, that if the mixture be very wet to start with (say, consisting of 60 per cent of water and only 40 per cent by weight of steam), adiabatic expansion will produce a dryness, some of the water present becoming re-evaporated at the expense of the sensible heat in the hot water. This would be shown by the ratio  $\frac{A F}{A D}$  being greater than  $\frac{B E}{B C}$ .

#### CHAPTER IV.—HEAT LOSSES.

#### 17. EFFECT OF STEAM JACKETING.

In all steam engines a comparatively large proportion of the steam which enters the cylinder is condensed immediately upon its admission, and its work lost to the engine for that period of the stroke. A part of the heat represented by this condensation is returned to the steam towards the end of the expansion period, when its capacity for doing work is considerably diminished; or it may only be returned during the exhaust stroke, when (except in the case of more than one cylinder) it is not only of no use, but impedes the free discharge of the exhaust by increasing the volume of the steam. Various methods have been adopted to reduce this initial condensation to a minimum, the principal of which are steam jacketing, superheating, compounding, and high speed. The various effects of these on the internal working steam of the engine, as shown by the  $\theta \phi$  diagram. form a very interesting and instructive study.

The effect of steam jacketing is very clearly shown by drawing the  $\theta \phi$  diagrams for the same engine, working with and without steam in the jackets. Figs. 16 and 17 are the diagrams for two trials of a triple-expansion engine at the Wapping Pumping Station of the London Hydraulic Power Company, made by Mr. Bryan Donkin in 1892, full particulars of which will be found in the third Report of the Steam Jacket Research Committee of the Institution of Mechanical Engineers, already referred to (see Proceedings of the Institution of Mechanical Engineers, October, 1894, page 536). The cylinders of the engine are 15 in., 22 in., and 36 in. diameter, with a stroke of 2 ft., with pistons coupled direct to water plungers 5 in. diameter. The steam pressure was about 135 lb. absolute, the speed about 56 revolutions per minute, and the total I.H.P. about 175. Comparing the two trials c and d, the former with boiler steam in all the three barrel jackets (the covers are not jacketed), and the latter



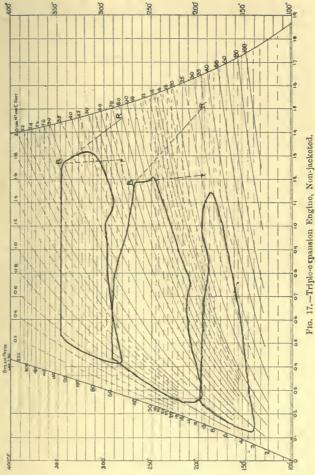


Fig. 17.-Triple-expansion Engine, Non-jacketed.

without steam in the jackets, the consumption of steam per I.H.P. per hour was 15.14 lb. and 17.17 lb. respectively, or 11.8 per cent less in the trial with steam in the jackets.

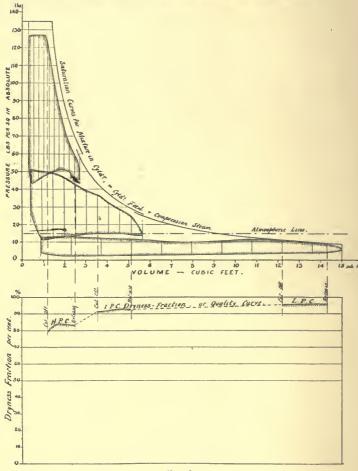


Fig. 18.

The mean indicator diagrams for the two trials are shown in figs. 18 and 19, the former with steam in all jackets and the latter without. The saturation curves for

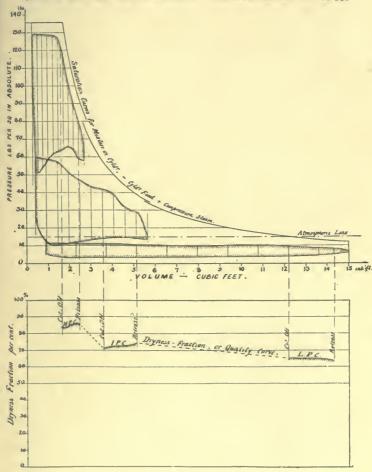


Fig. 19.

the known weights of the mixture of steam and water in each cylinder per stroke have been plotted, in order to show the comparative work losses in the two cases. These lines do not form one continuous curve, because the weight of compression or "dead steam" left in the cylinder at the end of each stroke is different for each cylinder: but it is more correct to plot the weight of the mixture than to plot the "live steam" admitted per stroke only, as the comparison of the actual indicator diagram with the former gives the amount of steam present at all points of the stroke. The saturation curves for 1 lb. of dry saturated steam can be plotted if the volumes be taken as the  $\theta \phi$  volumes already tabulated to construct the  $\theta \phi$ diagram, the table of densities given in the appendix being used for giving the pressures. The curves of dryness fraction shown underneath the mean indicator diagrams have been taken directly from the  $\theta \phi$  diagrams, figs. 16 and 17, and show very clearly the gradual condensation and re-evaporation during the progress of the steam from the high-pressure cylinder to the condenser. At certain periods of the stroke both condensation and re-evaporation may be taking place at the same time; in which case, the curves only show the excess of the one action over the other. The  $\theta \phi$  diagrams for the trial "c," with steam in the jackets, are shown in fig. 16, and those for trial "d," without steam in jackets, in fig. 17. In comparing the two diagrams, the most noticeable difference is the enormously-increased area of those in fig. 17, especially in the case of the intermediate and low-pressure cylinders. The reason for this is, the water formed during admission to the high-pressure cylinder is gradually re-evaporated by the live steam in the jackets during its passage through the three cylinders, until, when it leaves the low-pressure cylinder for the condenser, it consists of 96 per cent dry steam and 4 per cent of water in the trial with all the jackets on, as compared with 65 per cent dry steam and 35 per cent water in the non-jacketed trial.

The comparative areas of the diagrams in figs. 16 and 17 are given in Table VIII., which gives the areas of the  $\theta$   $\phi$  diagrams, expressed in B.T.U. per stroke, as compared with the actual I.H.P. for each cylinder, also expressed in B.T.U. per stroke. The mechanical equivalent of heat (J) has been taken as 772 foot-pounds. The volume factor is found by dividing the specific volume of 1 lb. of dry saturated steam, at the pressure shown where the dryness fraction has been calculated, by the actual volume occupied by the steam at that moment. For example, at release in the high-pressure cylinder of trial C, the pressure shown by the indicator diagrams was 59·3 lb. absolute, and the specific volume of dry steam at this pressure is

$$\frac{1}{0.14064} = 7.11$$
 cubic feet.

Multiply this by 0.855, the known dryness of the steam at that point, and the result is 6.08 cubic feet per pound occupied by the steam. But the volume of the high-pressure cylinder at release is 2.455 cubic feet; therefore the volume factor for the high-pressure cylinder in trial C will be

$$\frac{608}{2.455} = 2.476.$$

The volume factor may be looked upon as the reciprocal of the weight of dry steam passing through the cylinder, and the area given in column three of Table VIII. will be the work done in B.T.U. per stroke by 1 lb. weight of steam, which, divided by the volume factor in column 4, gives the work done in B.T.U. per stroke for the known weight of dry steam present in each cylinder. The difference between the figures given in columns 5 and 6 is accounted for by drawing the  $\theta$  diagrams in figs. 16 and 17 from a mean indicator diagram which was constructed from nine sets of actual indicator diagrams, taking one set per hour; whereas the figures given in column 6 are calculated from the actual I.H.P. developed in each cylinder, taken from all the indicator diagrams (some forty sets).

TABLE VIII.—Area of  $\theta \phi$  Diagrams for Triple-expansion Engine, with and without Steam in Jackets.

		Area of		Heat utilised per stroke.		
Trial and conditions.	Cylinder. $\theta \phi$ diagrams in B.T.U measure by planimet		Volume factor for each cylinder.	Calculated from $\theta \phi$ diagrams.	Calculated from ordinary indicator diagrams.	
m (	н Р.С	51.0	2.47	20.5	20.8	
Trial C.	ſ.P.C	68.1	2.865	23.8	23.05	
Steam in all	L, P, C	58.3	2.82	20.7	20.45	
jackets.	Total	167 4		65.0	64.3	
T	H.P.C	43.5	1.835	23.8	23.6	
Trial D.	I.P.C	58:45	1.985	29.4	29.4	
No steam in	L.P.C	34.52	2.06	16:7	16.3	
jackets.	Total	136 47		69.9	69.3	

# 18. THEORETICAL ENTROPY DIAGRAM FOR SUPERHEATED

When saturated steam is removed from the water from which it is generated, and heated beyond the temperature that corresponds to its pressure, it becomes superheated, the additional amount of heat received being

$$Q_1 = C p \times (\tau_1 - \tau);$$

where Q<sub>1</sub> = heat received as superheat per pound of steam;

Cp = specific heat of superheated steam at constantpressure (usually taken as 0.48);

 $\tau_1$  = temperature after superheating;

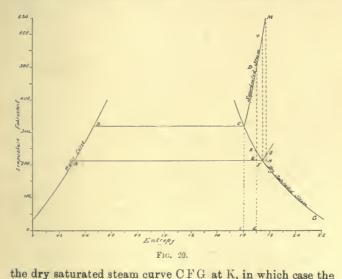
 $au = ext{temperature}$  of saturated steam of the same pressure.

Its entropy is therefore greater than that of corresponding saturated steam, and is equal to

$$\phi_s + 0.48 (\log_e \tau_1 - \log_e \tau),$$

 $\phi_{\delta}$  being the entropy of 1 lb. of dry saturated steam of the

same pressure (see Table II., page 13). This increased heat is represented on the entropy diagram, fig. 20, by the extension curve CD, the area under which CDdc (extended to the absolute zero of temperature) is equal to  $Q_1$ , or 0.48  $(\tau_1 - \tau)$ , where C is taken at temperature  $\tau$ , and D at  $\tau_1$ . If expansion be assumed adiabatic—that is, with a nonconducting cylinder and piston—the diagram is completed by the vertical line DE from  $\tau_1$  to  $\tau_2$ , which may intersect



steam will pass from the superheated to the saturated condition at K, and at the end of expansion be wet steam, with a dryness of  $\frac{AE}{AF}$ . In order to have dry steam at F, the steam must be superheated up to the point H, the latter being found by drawing the vertical F H to intersect the superheated steam curve CDM. If the superheating be carried beyond this point, as, for instance, to M, the steam remains superheated throughout the whole of the expansion

period (assumed adiabatic), and at release it still possesses an amount of superheat represented on the temperature scale by the amount NO, the curve FO being drawn by the same equation as CDM. Table IX. gives the entropy for 1 lb. of superheated steam, starting with dry saturated steam

TABLE IX.—ENTROPY OF SUPERHEATED STEAM FROM DRY SATURATED STEAM AT 320 DEG. FAH.

Temperature.		Amount of	Increase of	Total entropy above water at	
Deg. Fah.	Absolute. $ au_{\cdot}$	superheat. Deg. Fah.	entropy. $d \phi$ .	32 deg. Fah. <i>φ</i> .	
320	780	0	0	1.6043	
330	790	10	0.0061	1.6104	
340	800	20	0.0121	1.6164	
350	810	30	0.0181	1.6224	
360	820	40	0.0240	1.6283	
370	830	50	0.0299	1.6342	
380	840	60	0.0356	1.6399	
390	850	70	0.0413	1.6456	
400	860	80	0.0469	1.6512	
410	870	90	0.0524	1.6567	
420	880	100	0.0579	1.6622	
430	890	110	0.0634	1.6677	
440	900	120	0.0687	1.6730	
450	910	130	0.0740	1.6783	
460	920	140	0.0792	1.6835	
470	930	150	0.0844	1.6887	
480	940	160	0.0896	1.6939	
490	950	170	0.0947	1.6990	
500	960	180	0.0997	1.7040	
510	970	190	0.1046	1.7089	
520	980	200	0.1096	1.7139	

at 320 deg. Fah. (corresponding to about 90 lb. absolute pressure), for every 10 deg. up to 520 deg. Fah., or 200 deg. of superheat.

### 19. EFFECT OF SUPERHEATING.

The effect of superheating the steam, as shown by the  $\theta \phi$  diagram, has been very ably treated by Prof. Ripper in his recent paper read before the Institution of Civil Engineers (see Proceedings, vol. exxviii., January, 1897), and partially reproduced in *The Practical Engineer* (see vol. xvi., page 112).

The main object of using superheated steam is to minimise initial condensation by reducing the exchange of heat between the working steam and the metal walls. When using ordinary saturated steam, which usually contains a small quantity of suspended moisture, condensation during admission must produce a deposit of water upon all the clearance surfaces, which is re-evaporated as the temperature of the steam falls (principally during exhaust), at the expense of the heat stored up in the cylinder walls. The result is, the inner skin of the wall in the clearance surfaces is cooled considerably with every stroke, and condenses a comparatively large proportion of the incoming steam at the next stroke. With superheated steam, the heating up of the walls during admission is done at the expense of the superheat (providing that is sufficiently high); and at cut-off there should still be a little superheat left to evaporate the condensation formed by work being done, so that the steam leaves the cylinder at release just dry, but not superheated. The walls will then be absolutely dry during the exhaust stroke, and consequently require very little heating up at the commencement of the next stroke.

The comparative effect of jacketing and superheating is shown by the four  $\theta$   $\phi$  diagrams superposed in fig. 21, which represent four different trials made by Mr. Bryan Donkin on a small vertical engine (see Proceedings of the Institution of Mechanical Engineers, January, 1895, page 132). The

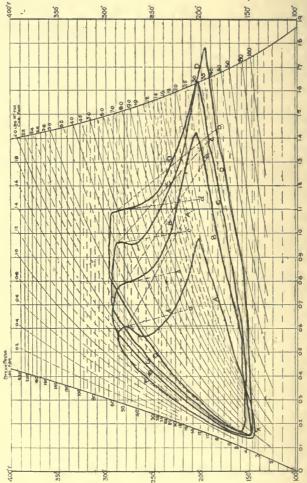


Fig. 21.—φ θ Disgrams for same Engine, Jacketed and Non-Jacketed, Saturated and Superheated Steam.

conditions and results of the four trials are shown by the following Table X.

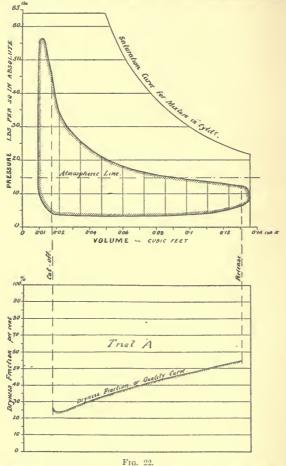
TABLE X.

linder. Remarks.	Steam in cylinder.	Steam in jackets,	Trial.
	Saturated steam	No steam in jackets	A
	Superheated steam	No steam in jackets	В
steam   27.25   cut-off, same	Saturated steam	Saturated steam in jackets	C
	Superheated steam	Superheated steam in jackets	D
steam 45.6 steam 28.4 steam 27.25 steam 27.25	Superheated steam Saturated steam	No steam in jackets Saturated steam in jackets	В

The mean indicator diagrams for the four trials are shown in figs. 22 to 25, inclusive, and form a very instructive object lesson to steam users, as showing the different amounts of work which 1 lb. of steam can be made to do if the conditions under which it is admitted to the cylinder are properly arranged. The weight of steam used per I.H.P. per hour, as given in table, is not an exact comparison when using superheated steam, because 1 lb, of the latter contains more heat than 1 lb. of saturated steam of the same pressure. A better standard of comparison is that recommended by the Thermal Efficiencies Committee of the Institution of Civil Engineers, viz., the number of British thermal units of heat used per I.H.P. per minute, taking the total heat contained in the steam on the boiler side of the stop valve, less the sensible heat contained in the steam (as water) in the exhaust pipe. This quantity of heat, for the four trials A, B, C, D, is 813, 500, 493, and 368 B.T.U.'s per I.H.P. respectively, and they represent the heat used per minute in producing one I.H.P., independently of the pressure or condition of the steam.

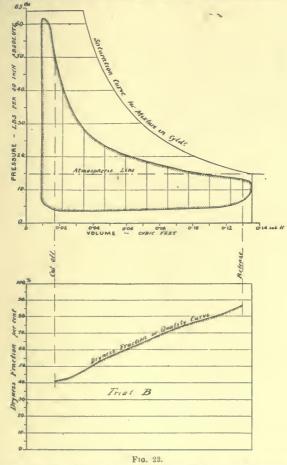
The results of the four trials are shown graphically by their  $\theta \phi$  diagrams in fig. 21. The smallness of the diagram in trial A is most noticeable, and shows the enormous loss by condensation during admission which is always produced in

non-jacketed cylinders, especially small ones. The partial re-evaporation during expansion, at the expense of the heat



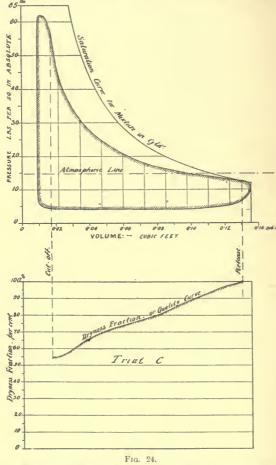
in the walls, tends to counterbalance the initial loss; but even at release the weight of steam present in the cylinder

is only about one-half what it should have been. The amount of the loss in this particular case was very much increased



by the cut-off taking place very early (at one-sixteenth of the stroke), and the cylinder being so very small (6 in.

diameter, 8 in. stroke); so that the clearance surface, and surface of cylinder exposed to steam during admission, per



pound of steam present, would be very high indeed with a small engine working under these conditions. The conden-

sation in a larger engine would be relatively much smaller, as shown by figs. 11 and 17.

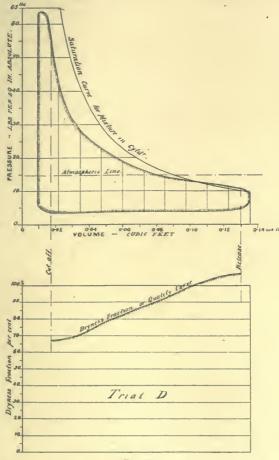


Fig. 25.

In trial B, the diagrams for which are given in fig. 23, with the steam superheated only 34 deg. Fah., and still

without steam in the jackets, the reduction effected in the initial condensation is most marked. Instead of there being only 24 per cent of the steam which was admitted present in the cylinder at cut-off, as in trial A, there is now 41 per cent, and the drier walls have produced a larger amount of re-evaporation.

In trial C (see fig. 24), with saturated steam in the cylinder and jackets, there is an initial dryness of 0.53 at cut-off, and the expansion is continued with a gradually-increasing dryness of steam, until at release all the water has been re-evaporated, and it is discharged to the condenser as all dry steam.

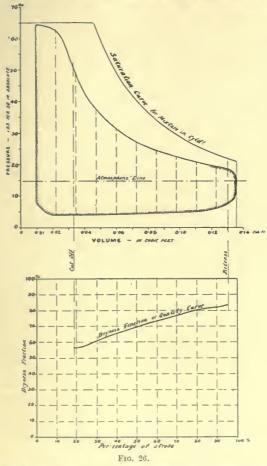
In trial D, fig. 25, using steam superheated 59 deg. Fah. in both the cylinder and the jackets, all the water has disappeared by the time the piston has made 0.8 of its stroke, after which the curve crosses the dry-steam boundary, and is slightly superheated. The "toe" of the diagram in trial D has been calculated by assuming that superheated steam follows the law of expansion of a perfect gas, viz., that the product of the pressure and volume varies directly as the absolute temperature, or

$$\frac{p \cdot v}{\tau} = a \text{ constant.}$$

## 20. Effect of Speed.

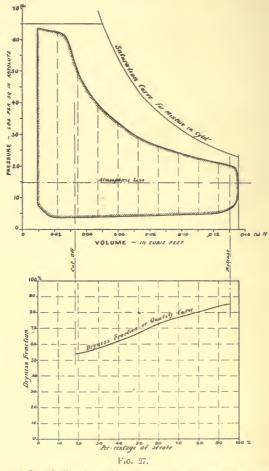
The effect of increased speed of rotation on the  $\theta \phi$  diagram is not very marked. Fig. 28 shows the diagram for two trials on the same engine, at the same steam pressure, cutoff, and general conditions, but in the first trial (shown by the full line in fig. 28) the speed was 216 revolutions per minute, and in the second (shown dotted) at 115 revolutions, or about one-half. With the increased speed there is a little less initial condensation, but the re-evaporation during expansion is greater in the half-speed trial, probably because the time taken by the engine to complete any given portion of its cycle is longer in the latter case, and allows the jackets to make their influence more effective. These

diagrams are taken from Mr. Bryan Donkin's experiment, Nos. 121 and 122 (see Proceedings of the Institution of



Mechanical Engineers, January, 1895). The mean indicator diagrams and quality curves for the two trials are shown in

figs. 26 and 27; the former at full speed, and the latter at half speed.



The trials of the late Mr. P. W. Willans on his centralvalve engine, recorded in the Proceedings of the Institution of Civil Engineers (vol. xciii., 1888, and vol. cxiv., 1893), prove that the percentage of steam not accounted for by the indicator diagram at cut-off varies inversely as the square root of the number of revolutions per minute. In fact, the amount of initial condensation can be approximately calculated by the following formula:—

$$\frac{W}{I} = c \frac{\log_e r}{D \times \sqrt{N}}$$

where W = the weight of steam per hour not accounted for by the indicator at cut-off;

> I = the weight of steam per hour accounted for by the indicator at cut-off;

r =number of expansions;

D = diameter of cylinder in feet;

N = revolutions of engine per minute;

c = a numerical coefficient, depending on the design of cylinders and conditions of working.

For ordinary slide-valve engines, with an average amount of clearance surface, c can be taken at about 3 or 4 for jacketed cylinders and 6 for non-jacketed cylinders.

The value of the coefficient c for the two trials just quoted (see figs. 26 and 27) is shown by the accompanying Table XI. to be 2.07 at full speed, and 2.11 at half speed.

TABLE XI.—Condensation at Two Speeds: Other Conpitions Remaining Constant.

Revs. per minute.	Indicated steam at cut-off I, Pounds per hour.	Actual steam used.	Steam condensed during admission.		ation
		from experiment. A. Pounds per hour.	W. Pounds perhour.	Ratio W I	Condensation coefficient
216-3	130-1	179-4	49*8	0.379	2.07
114.9	68.0	104-1	36.1	0.531	2.11

#### 21. Compounding.

The use of high-pressure steam, requiring a greater number of expansions for economical working, has necessitated the

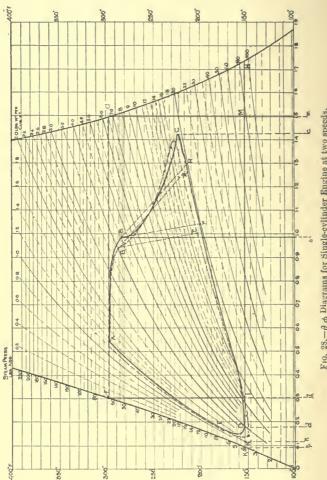


Fig. 28.— $\theta$   $\phi$  Diagrams for Single-cylinder Engine at two speeds.

introduction of multi-cylinder engines, in order that the range of temperature in each cylinder shall not be too large. The question therefore arises, What is the most economical number of expansions—i.e., what is the point beyond which the extra work gained per pound of steam by expansive working is more than neutralised by the loss due to initial condensation? The late Mr. Willans' trials seemed to point to the law—

$$r = \frac{p + 10}{25}$$
;

where r = most economical number of expansions;

p = initial steam pressure (absolute) in pounds per square inch,

for high-speed non-condensing engines; but the author thinks that with moderate-sized engines, well jacketed, the expansions may be greater than those given by the above law, and would substitute the following rule,

$$r = \frac{p + 20}{20},$$
$$r = 1 + \frac{p}{20},$$

or

as being more in accordance with experimental results obtained with different engines. The enormous loss of heat which accompanies very early cut-off is clearly seen by comparing figs. 31 and 32. The diagrams given in fig. 31 are for a compound non-jacketed engine of about 400 I.H.P., tested by Mr. Michael Longridge on October 21st, 1896, the full particulars of which are recorded in the Report of the Engine, Boiler, and Employers' Liability Association for 1896; partly re-published in The Practical Engineer for October 15th, 1897. The diagrams indicate a fairly economical performance, as they almost fill up the available area between the steam and water boundary curves. Comparing them with the  $\theta \phi$  diagram given in fig. 32 for a compound non-jacketed engine of the same type, and indicating about the same power, but which was made much too large for the

work required, it will be seen how the automatic expansion controlled by the governor destroyed the efficiency of the engine in the latter case (fig. 32). The mean indicator diagrams and quality curves for the two trials are shown in figs. 29 and 30. The cut-off in the high-pressure cylinder for Mr. Longridge's trial was 0.31 of the stroke;

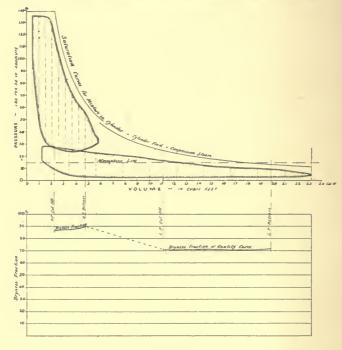


Fig. 29.

and for the diagrams given in fig. 32 the mean cutoff in the high-pressure cylinder was 0.054 of the stroke. The result is, that the steam consumption is increased from 13.9 lb. per I.H.P. per hour in the former, to 27.2 lb. in the latter case. The importance of this waste is realised when one considers that a difference of 13 lb. of steam per I.H.P. per hour for an engine of this size, running night and day, represents a loss of some £15 a week in the coal bill alone.

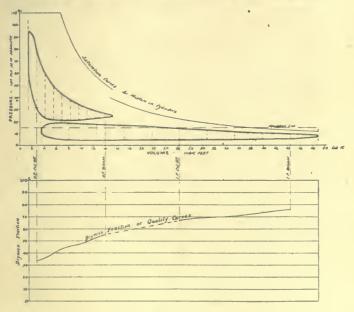
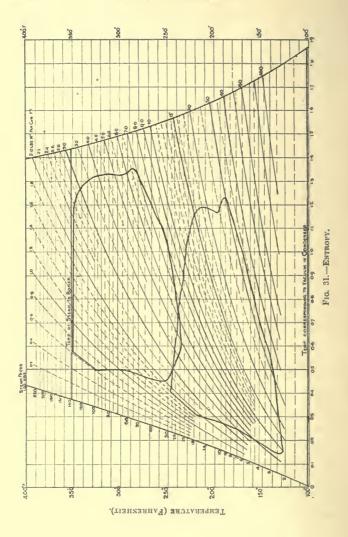
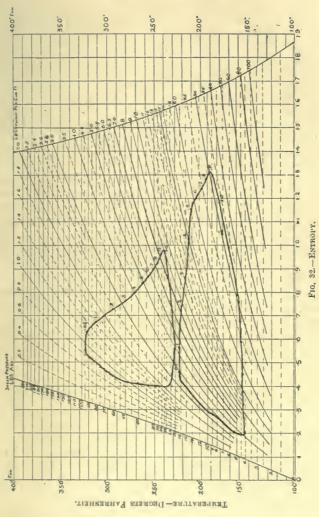


Fig. 30.

## 22. INITIAL CONDENSATION.

The weight of steam missing at cut-off may be accounted for in two ways. In the first place, it may be assumed to have entered the cylinder as hot water—i.e., due to priming and condensation in pipes; in which case it would part with its heat during expansion by evaporating a very small portion of the water formed by adiabatic expansion. Assuming a non-conducting cylinder, the expansion curve would then theoretically follow the dotted line Br, shown





6T

on the high-pressure and intermediate cylinder diagrams in figs. 16, 17, &c. This line was called by the late Mr. Willans the "priming water heat-recovery curve" (see discussion by Capt. Sankey of Prof. Beare's paper on "Marine-engine Trials"-Proceedings, Institution of Mechanical Engineers, February, 1894). On the other hand, the quantity of steam missing at cut-off may be assumed to have entered the cylinder as dry steam, and been condensed during admission; in all probability the larger portion of it would be condensed before the stroke had commenced. Assuming the admission valves to be tight, and the cylinder non-conducting, the latent heat of this condensed steam will be stored up in the cylinder walls for use at a later period. If all the heat be returned to the steam during expansion, the  $\theta \phi$  diagram from the point of cut-off should follow the dotted line BR, shown in figs. 16 and 17. This line is called the "condensation water heat-recovery line." The actual expansion curve of the engine should lie somewhere between these two extreme cases, assuming no external heat to be furnished from a jacket or similar source; in which case the position of the actual curve relatively to Br and BR will show the nature and extent of heat exchange from the metal to the steam after cut-off. Should the cylinder be of small dimensions, or the cut-off very early, this exchange of heat becomes a considerable factor in determining where all the heat of the steam has gone to.

### METHOD OF DRAWING Br AND BR LINES.

Let x represent the fractional dryness of the steam at cut-off, scaled from the  $\theta \phi$  diagram; then (1-x) will be the weight of water in pounds present in the cylinder, as the  $\theta \phi$  diagram is drawn for 1 lb. of the mixture. But the heat which (1-x) pounds of water will evolve in cooling from a temperature  $\tau_1$  to a temperature  $\tau$  is

$$(1-x).s.(\tau_1-\tau),$$

where s is the mean specific heat of water between  $\tau$  and  $\tau_1$ 

Let  $\tau_1$  equal the temperature of the steam at cut-off, and  $\tau$  the minimum temperature of steam in the cylinder. Draw horizontal lines across the  $\theta$   $\phi$  chart at these temperatures (see AC and DE, fig. 16), and drop a vertical ordinate from A, meeting DE in F, as shown. Then the triangle DAF, extended to the absolute zero of temperature, represents the heat given to 1 lb. of water in raising it from  $\tau$  to  $\tau_1$ ; and vice versa, the heat evolved by 1 lb. of water in cooling from  $\tau_1$  to  $\tau$ . Divide DF at G, in the proportion of the

dryness fraction at cut-off—i.e., make  $\frac{D\,G}{D\,F}$  equal to  $\frac{A\,B}{A\,C}$ ; then

the triangle GAF, extended to the zero of temperature, will represent the heat given up by (1-x) pounds of water in cooling from  $\tau_1$  to  $\tau$ . This is not mathematically correct, because the curve DA is not quite a straight line; but the error introduced in assuming DA straight is quite negligible, and does not materially affect the length of the line GF. Now turn to the point of cut-off, B, and drop a vertical ordinate from B meeting DE in H, and make the triangle BHK equal to the triangle AGF. The line BH represents adiabatic expansion, assuming a non-conducting cylinder; and the line BK (or Br as it is usually called) will represent the theoretical expansion curve, assuming the (1-x) pounds of water present at B to give up its sensible

heat to the steam as the temperature falls; also  $\frac{H\;K}{D\;E}$  will

represent, to scale, the fractional weight of water reevaporated by the heat received from the hot water.

To draw the BR line of condensation water-heat recovery, multiply the weight of water present at cut-off by the latent heat of steam at the same temperature, and divide the product by the mean absolute temperature of the cycle. This will give the amount of increased entropy which would be added to that possessed by the steam present at cut-off, should all the heat of condensation be returned by the walls during expansion.

Using the same notation as before, the increase of entropy will be—

$$\phi_1-\phi=\frac{(1-x)\,.\,\mathbf{L}_1}{T_m}\;;$$

where  $L_1$  is the latent heat of 1 lb. steam at  $\tau_1$ ;

T<sub>m</sub> is the mean absolute temperature between cut-off and exhaust:

 $\phi_1 - \phi$  is the increase in entropy obtained from the re-evaporation.

Referring again to fig. 16, make H R equal to  $(\phi_1 - \phi)$ , to the scale of entropy used, and join B R, which will represent the theoretical expansion curve, assuming the walls to give back to the steam during expansion all the heat they received from the liquefaction of (1-x) pounds of steam up to cut-off.

The lines Br and BR are shown on the various diagrams, so that the position of the actual expansion line relative to the two theoretical lines can be noted.

# 23. Measurement of Heat Losses by the Aid of the $\theta \phi$ Diagram.

THE most important application of the  $\theta \phi$  diagram is, that all the various losses and exchanges of heat can be easily represented graphically, and to a fixed scale of heat units. Referring to fig. 28, and considering only the trial at full speed (shown by the full line), the area of the diagram ABCDE represents in heat units the net work done on the piston per stroke. Draw the maximum and minimum steam temperature lines, FG and KH respectively, and drop the vertical ordinate G M, meeting K H in M; then, if k and m be the ordinates K and M respectively projected on the absolute zero of temperature, the area kKFGMm will represent to scale the total heat supplied to the steam used per stroke, assuming the feed water to be returned to the boiler at the temperature denoted by the line KH. The weight of steam used per stroke is assumed to be 1 lb., for simplicity of calculation; but in reality it is

pounds per stroke, as explained previously. The ratio of the two areas

ABCDE

will therefore represent the absolute thermal efficiency of the engine under these conditions. If the engine consists of more than one cylinder, the line FG should be drawn at a temperature corresponding to the pressure in the steam pipe near the high-pressure cylinder, and the line KH at the temperature of the hot-well discharge from the condenser. The work done will also be the sum of the areas of the diagrams for each cylinder.

The work done by a perfect heat engine working between the same limits of temperature can be represented by the rectangle L F G M—i.e., the Carnot reversible cycle with adiabatic expansion and compression, and no clearance. The efficiency of the perfect heat engine will therefore be represented on the  $\theta$   $\phi$  diagram by the ratio of the two areas

LFGM ;

or

The Rankine cycle, which some engineering professors now prefer to use as the standard steam engine of comparison (see report of the Thermal Efficiency Committee of the Institution of Civil Engineers, 1898), includes the heating up of the feed water from the temperature of the exhaust, and is represented in fig. 28 by the area K F G M; the upper limit of temperature, F G, being that corresponding to the pressure of the steam on the boiler side of the stop valve, and the lower limit of temperature, K M, being that of the steam in the exhaust pipe, near the engine. If superheated steam be used, the extra work theoretically obtained should be included, as shown in fig. 20 (page 47). The efficiency of the standard engine of comparison (Rankine cycle) is therefore the ratio of

KFGM kKFGMm. Consequently, the ratio of the actual thermal efficiency to that of the standard heat engine (Rankine cycle) will be represented by

ABCDE KFGM

The ratio of the actual thermal efficiency to that of a Carnot engine working between the same limits of temperature is  $\frac{A \ B \ C \ D \ E}{L \ F \ G \ M} .$ 

Mr. Willans adopted a fixed minimum pressure for condensing engines, corresponding to a temperature of 110 deg. Fah., in calculating the thermal efficiency of his central-valve engine in his trials, published by the Institution of Civil Engineers (see vol. cxiv., p. 9, Proc. Inst. C. E.), and called the thermal efficiency of the engine the ratio of the areas

ABCDE, KFGM,

with FG drawn to correspond to the boiler pressure, and KG drawn at 1.267 lb., corresponding to 110 deg. Fah. This represents the thermal efficiency of the engine compared with a perfect steam engine, without clearance, receiving steam at boiler pressure and expanding it adiabatically completely down to the condenser pressure of 1.267 lb., and discharged at that pressure. Mr. Willans' argument was that it was unfair to credit a steam engine with the heat contained by the steam at a temperature (110 deg. Fah.) below which it was useless for power purposes. He therefore compared his engine with a perfect steam engine, and obtained a thermal efficiency of some 50 to 60 per cent; whereas the absolute thermal efficiency of steam engines, or the ratio of work done to heat supplied, is only about 12 to 15 per cent.

For the particular trial under consideration, the diagram of which is given in fig. 28, the areas measured by a planimeter gave the following results: Area ABCDE, equal to the net work done per stroke,

= 94.85 B.T.U. for 1 lb. steam.

Area  $k \times F \times M = \text{heat received per stroke} = 1050^{\circ}2 \text{ B.T.U.}$ Thermal efficiency, or ratio of work done to heat supplied,

$$=\frac{94.85}{1050.2}=0.0903.$$

Area LFGM = work done by a perfect heat engine working between same limits of temperature on the Carnot cycle = 1716 B.T.U. Thermal efficiency of perfect heat engine (Carnot cycle)—

$$\frac{\tau - \tau_1}{\tau} = \frac{171.6}{902.2} = 0.190.$$

Ratio of actual thermal efficiency to thermal efficiency of perfect heat engine (Carnot cycle)

$$=\frac{94.85}{171.6}=0.553.$$

Work done by perfect steam engine (Willans' method), expanding done to 110 deg. Fah., = 249 6 B.T.U.

Ratio of actual thermal efficiency to the thermal efficiency of a perfect steam engine (Willams' standard)

$$=\frac{94.85}{249.6}=0.380.$$

The difference is thus seen between the various methods of expressing the thermal efficiency of a steam engine, according to the way in which the term "efficiency" is used and understood.

The areas representing the heat losses in fig. 18 are:

1. Loss due to clearance, equal to area  $k \text{ KFA} \to N n^*$  (extended to absolute zero of temperature), = 30.0 B.T.U.

- 2. Loss by heat given to walls during admission = area  $b \ B \ A \ G \ M \ m$ ; less the heat given by the walls to the steam during expansion, represented by  $b \ B \ C \ c = 387.7 304.7 = 83.0 \ B.T.U.$
- 3. Heat carried away by exhaust steam = area  $d \, \mathrm{D} \, \mathrm{C} \, c$ ; plus compression  $n \, \mathrm{N} \, \mathrm{E} \, \mathrm{D} \, d$

$$= 801.8 + 40.6 = 842.4 \text{ B.T.U.}$$

The italic letters k, m, n, &c., are the extremities of the ordinates K, M, N, &c., on the line of absolute zero of temperature (- 400 deg. Fab.)

The sum of these quantities (1+2+3), plus the heat represented by the useful work done (area ABCDE = 94.85 B.T.U.), should be equal to the total heat received by the engine, represented by the area  $k \times FGMm$  (1,050.2 B.T.U.)

Accounting for all the heat in a balance sheet, according to Hirn's method of treatment, the quantities expressed in B.T.U.'s per stroke for 1 lb. of steam are:—

#### HEAT BALANCE SHEET.

Heat received.	To-		Hec	it accounte	d for.	
By steam received. Area $k \times F \times G \times M m$ . = 1050'2 B.T.U. = 100 per cent.	1. Loss by	ork clearance walls c shaust steam	30.0	B.T.U. = B.T.U. =	2.86 pe 7.90 pe	r cent
		Total =	1050.25	B.T.U. =	100 u pe	rcen

## CHAPTER V.—APPLICATION TO THE GAS ENGINE.

## 24. General Considerations Applicable to all Permanent Gases.

If 1 lb. of any gas be heated through a small rise of temperature denoted by dt, the amount of heat received can be calculated from its increased energy by the usual formula:

$$dQ = C v \times dt + A \times P \times dv \dots \dots (1)$$

where

dQ = the amount of heat received in B.T.U.;

Cv =the specific heat of the gas at constant volume ;

 $A = \frac{1}{J}$ , or the reciprocal of Joule's mechanical equivalent of heat;

P = thepressure of the gas in lbs. per square foot;

dv = the increase of volume in cubic feet.

The first part of the equation  $(Cv \times dt)$  represents the additional internal energy of the gas; and the second part  $(A \times P \times dv)$  that corresponding to the external resistance overcome.

But the combination of Boyle's and Charles's laws proves that the product of pressure and volume of any permanent gas varies directly as the absolute temperature; or

$$P \times V = R \times \tau \quad . \quad . \quad . \quad . \quad (2)$$

where V = volume of the gas in cubic feet;

R =some numerical constant (Kp - Kv);

 $\tau$  = absolute temperature.

Substituting the value of P so found in formula (1), we get—

$$d Q = (C v \times d t) + \left(A \times R \times \tau \times \frac{d v}{V}\right) . . (3)$$

and, dividing all through by 7,

$$\frac{dQ}{\tau} = C v \times \frac{dt}{\tau} + A \cdot \times R \cdot \times \frac{dv}{V} \cdot \dots (4)$$

But  $\frac{d}{\tau}Q$  is the required small increase of entropy corresponding to the increase of temperature d t.

The external work done by 1 lb. of gas in heating

= 
$$P \cdot dv = dt (Cp - Cv) J (in work units);$$

therefore  $Cp - Cv = A \cdot P \cdot \frac{dv}{dt}$ ,

but also  $Cp - Cv = A \cdot R \cdot ;$ 

and, substituting this value for A. R in equation (4), we get

$$\frac{d \mathbf{Q}}{\tau} \text{ or } d \phi = \mathbf{C} v \frac{d t}{\tau} + (\mathbf{C} p - \mathbf{C} v) \frac{d v}{\mathbf{V}} . \quad . \quad (5)$$

This is the general formula from which the change of entropy of a gas can be calculated under the various conditions of change. For example, during explosion in a gas engine the volume is constant, or very nearly so, and therefore dv = 0. Under these circumstances,

$$d \phi = \mathbf{C} v \cdot \frac{d t}{\tau};$$

which, integrated, gives

$$d \phi = \int_{\tau_0}^{\tau_1} \mathbf{C} \, v \cdot \frac{d \, t}{\tau}$$
$$= \mathbf{C} \, v \cdot \log_{\ell} \frac{\tau_1}{\tau_0} \cdot \dots \quad (6)$$

On the other hand, if the pressure remains constant, as is sometimes the case just after explosion in a gas engine, the volume and the temperature increase in accordance with Charles's law, viz.,

$$\frac{d v}{V} = \frac{d t}{\tau}$$
, or  $C v \cdot \frac{d t}{\tau} = C v \cdot \frac{d v}{V}$ .

In this case, equation (5) becomes—

$$d \phi = C p \cdot \frac{d v}{V};$$

$$C p \cdot \frac{d t}{V};$$

or,

which, reduced,

$$= d \phi = C p \cdot \log_{\ell} \frac{\tau_1}{\tau_0} \cdot \cdot \cdot \cdot \cdot (7)$$

Where both the pressure and the volume of the gas change, as during expansion in a gas engine, the change of entropy must be calculated from the formula representing the law of the particular expansion or compression, viz.,

$$P \cdot V^k = P_1 V_1^k \cdot \dots \cdot (8)$$

But, from equation (2),

$$P \cdot V = R \cdot \tau$$

and, substituting the value of

$$P = \frac{R \cdot \tau}{V} \text{ in (8),}$$

we get

$$R.\tau.V^{k-1} = P_1 V_1^k ... (9)$$

or, taking the value of  $\frac{d v}{\overline{V}}$  from (3), and substituting in (5), we get

$$d \phi = C v \cdot \frac{k - \gamma}{k - 1} \cdot \frac{d t}{\tau} \cdot \dots$$
 (10)

which, integrated, gives-

$$\phi_1 - \phi_2 = \mathbf{C} \, v \cdot \frac{k - \gamma}{k - 1} \cdot \log_e \frac{\tau_1}{\tau_2},$$

where  $\phi_1$  = initial entropy;

 $\phi_2 = \text{final entropy};$ 

 $\tau_1 = \text{initial temperature};$ 

 $\tau_2$  = final temperature.

This is the general formula for finding the change in entropy in all changes represented by the law

$$P \cdot V^k = P_1 V_1^k.$$

Given any expansion or compression curve of an indicator diagram for a gas engine, k is found by measuring any two co-ordinates, thus:

$$P \cdot V^{k} = P_{1} V_{1}^{k};$$

$$\therefore k = \frac{\log P - \log P_{1}}{\log V_{1} - \log V}.$$

In the particular case where  $k = \gamma$ —that is to say, for

adiabatic expansion or compression—the change of entropy becomes nil, as  $k - \gamma = 0$ , and therefore

$$\phi_1 - \phi_2 = C v \times \frac{k - \gamma}{k - 1} \times \log_e \frac{\tau_2}{\tau_1} = 0.$$

The entropy diagram, therefore, for any adiabatic change, is a vertical straight line.

25.—Entropy Diagram for Theoretical Gas Engine, with Adiabatic Expansion and Compression.

The indicator diagram for a perfect gas engine, working on the Otto or Beau de Rochas cycle, is shown in fig. 33, with pressure as ordinates and volume as abscisse. In this diagram AB represents the compression of the charge, assumed to be adiabatic—that is, following the law  $p \cdot v^{\gamma} = \mathbf{a}$ 

constant, where  $\gamma$  is the ratio  $\frac{\mathrm{C}}{\mathrm{C}} \frac{p}{v}$  of the mean specific heats

of the gaseous mixture. BC shows the explosion, with instantaneous increase of pressure at constant volume, and CD the expansion period, also assumed to be adiabatic. DA represents the exhaust, and AE the discharge and suction strokes. The entropy diagram for this cycle is shown in fig. 34, and is drawn for 1 lb. of the gaseous mixture, the vertical ordinates representing absolute temperatures ( $\tau$ ) and the base, entropy or  $\phi$ . Starting with the mixture at A, it is compressed from volume V a to V b adiabatically, without receiving or rejecting any heat. The process is therefore represented in fig. 34 by the vertical line AB, the entropy at B being the same as at A. The pressure at B can be found by the general formulæ—

$$Pa \times Va^{\gamma} = Pb \times Vb^{\gamma};$$

or, taking  $\gamma = 1.4$  (its approximate value),

 $\log Pb = \log Pa + 1.4 \log Va - 1.4 \log Vb.$ 

The temperature at B is found from the general formulæ-

$$\tau b = \frac{Pb \times Vb}{Kv - Kv},$$

where  $\tau b$  = absolute temperature at B;

Pb = pressure at B in lbs. per square foot;

Vb = specific volume of the mixture at B;

Kp = specific heat of the mixture at constant pressure;

Kv =specific heat of the mixture at constant volume.

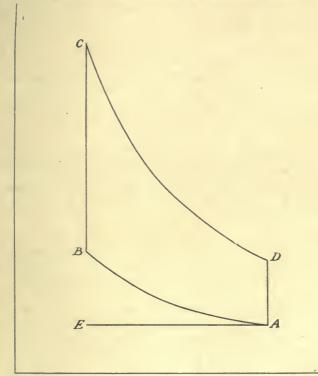


Fig. 33.

[Note,—K p and K v are in ft.-lb. units, = J  $\times$  C p and J  $\times$  C v]

where J = Joule's mechanical equivalent of heat = 772 ft.-lbs.

If the temperature of the mixture at A be known, and the ratio of the compression, the temperature at B can be calculated direct from the general formula for adiabatic compression of a gas—

 $\tau b = \tau a (r)^{\gamma - 1}$ 

where  $\tau b$  and  $\tau a$  are the absolute temperatures at B and A respectively,

 $r = \text{ratio of compression} = \frac{V \alpha}{V b}$ ;  $\gamma = \text{ratio of specific heats}.$ 

From B to C the pressure is increased instantaneously before the piston has moved, and therefore the increase of temperature will be directly proportional to the increase of pressure, or

 $\tau c = \frac{\tau b \times Pc}{Pb};$ 

Pc and Pb being the absolute pressures in lbs. per square foot at C and B respectively. When the pressure at C is not known, its theoretical temperature can be calculated from the calorific value of the gas, assuming perfect combustion; thus

 $\tau c - \tau b = \frac{H}{R \times C v}$ 

where H = total heat of combustion of 1 lb. of the particular gas used;

R = ratio (by weight) of the gas, air, and diluent to the gas.

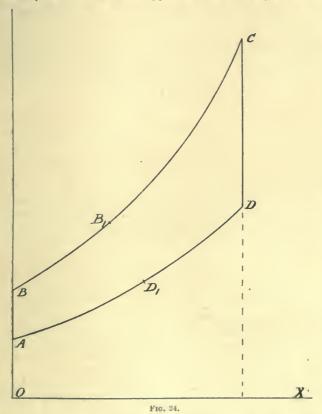
In actual practice, this theoretical rise of temperature is, for various reasons, never obtained. An average value of H for London lighting gas is about 19,000 B.T.U.'s per pound. The ratio R varies very considerably with different types of engines and conditions of working, but is about 20 to 30 for lighting gas. The increase of entropy during explosion will be represented in fig. 34 by the logarithmic curve BC, whose equation is  $\tau c$ 

 $d\phi = \int_{-1}^{\infty} \mathbf{C} v \cdot \frac{dt}{\tau};$ 

which, integrated, gives

$$\phi c - \phi b = C v \times \log_e \frac{\tau c}{\tau b};$$

that is, the increase of entropy from B to C is equal to the



specific heat of the mixture at constant volume (about 0.18 to 0.19), multiplied by the hyperbolic logarithm of the ratio of the two temperatures. Knowing  $\tau c$  and  $(\phi c - \phi b)$ , the

curve BC can be plotted as in fig. 34, all intermediate points such as  $B_1$  being calculated from the formula by substitution, thus—

 $\phi \dot{b}_1 - \phi b = \mathbf{C} v \times \log_e \frac{\tau b_1}{\tau b}.$ 

A short geometrical method of constructing such curves as B C and D A will be explained later.

The adiabatic expansion is denoted on the entropy diagram by the vertical line C D, fig. 34, the gas neither receiving nor evolving heat, and the temperature falling from  $\tau c$  to  $\tau d$ ; the latter being obtained from the formula,

$$\tau d = \frac{\operatorname{P} d \times \operatorname{V} d}{\operatorname{K} p - \operatorname{K} v},$$

and Pd being obtained from the usual formula,

$$Pd \times Vd^{\gamma} = Pc \times Vc^{\gamma};$$
  
 $Vd = Va$ , and  $Vc = Vb$ ,  
 $Pd \times Va^{\gamma} = Pc \times Vb^{\gamma};$ 

or, as

or

$$\log P d = \log P c + \gamma \cdot \log V b - \gamma \cdot \log V a$$

From D to A, or exhaust at constant volume, the entropy diagram again assumes a lcgarithmic curve D A, fig. 34, the temperature at A falling to its initial point, and its change of entropy being equal to

$$\phi d - \phi a = C v \times \log_{\ell} \frac{a}{\tau d}.$$

This change of entropy must always be negative, because  $\frac{\tau a}{\tau d}$  is less than unity—that is to say, the *increase* of entropy from D to A is a *minus* quantity or a loss, the gas giving up its heat to the exhaust. The entropy for any intermediate point D<sub>1</sub> on the curve D A can be calculated by substitution:

$$\phi \, d_1 - \phi \, d = \mathbf{C} \, v \times \log_{\ell} \frac{\tau \, d}{\tau \, d_1}.$$

The exhaust and suction strokes A E do not have any effect upon the entropy diagram, as the temperature during those strokes is assumed constant.

The diagram is completed by drawing O X at the absolute zero of temperature, when the work done per cycle will be equal to the area enclosed by ABCD, fig. 34, in heat units;

the heat received per cycle will be equal to the area O  $\rm B\,C\,X$  , and the thermal efficiency or

 $\frac{\text{work done}}{\text{heat received}} = \text{the ratio of the two areas } \frac{A B C D}{O B C X}.$ 

The heat given to the exhaust gases will be equal to the area O A D X.

It is evident from the entropy diagram that the two quantities  $(\phi c - \phi b)$  and  $(\phi d - \phi a)$  being equal, and the two curves BC and AD following the same law, the ratio of the two temperatures is a constant quantity, and depends entirely upon the amount of compression—that is,

$$\frac{\tau b}{\tau a} = \frac{\tau c}{\tau d},$$

and the higher this ratio, the higher will be the thermal efficiency.

26.—Entropy Diagram for Actual Gas-engine Trial.

For example, take the trial of a 7 horse power Crossley-Otto engine, made by Professor Capper, at King's College, London, on December 7th, 1892, full particulars of which are given by Mr. Bryan Donkin, in his book on "Gas, Air, and Oil Engines," from which the following data and mean indicator diagram (fig. 35), are taken.

The principal particulars of the trial are :-Cylinder, 81 in. diameter by 18 in. stroke. Revolutions per minute...... 162'5 Explosions 71.2 Net I.H.P. 13:32 Cylinder volume ..... 0.591 cubic foot. Clearance volume..... 0.2467 Total volume ..... 0.8377 Gas used (by meter) ...... 279.75 cb. ft. per hour. Gas used per explosion, at atmospheric pressure and temperature 0.06544 cubic foot. Gas used at pressure and temperature in cylinder at A ..... 0.0822Air used (from cylinder volume)... 0.7556 cubic foot 7T per explosion.

The mean indicator diagram, fig. 35, furnishes the following pressures and volumes:—

$$p_a = 13.8$$
 lb. per square inch absolute.  $p_b = 67.8$  , , , ,  $p_c = 240$  , , , ,  $p_d = 240$  , , , , ,  $p_d = 48.71$  , , , , ,  $v_a = 0.8377$  cubic foot.  $v_b = 0.2467$  , ,  $v_c = 0.2467$  , ,  $v_d = 0.2617$  , ,  $v_e = 0.8377$  . , .

The point E is taken on the ideal expansion curve, or the actual expansion line continued to the end of the stroke without exhausting. From the above pressures and volumes, the index k in the equation  $p_a \times v_a{}^k = p_b \times v_b{}^k$  is calculated to be—

For expansion, 
$$k = 1.3707$$
; , compression,  $k = 1.3022$ .

During compression, the mixture consists of air, gas, and exhaust products, in known proportions, and of known chemical analysis; therefore K p and K v for the mixture can be calculated in the same way as for any compound gas. The proportions of air, gas, and diluent present during compression are calculated thus:—

1. Gas, 0.06544 cubic foot per explosion at atmospheric pressure and temperature, with a specific volume of 34.87 cubic feet per pound,

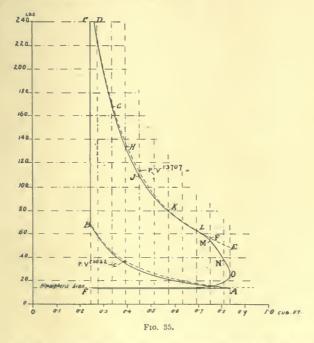
$$=\frac{0.06544}{34.87}=0.001877$$
 lb. per explosion.

2. Exhaust products left in cylinder at end of previous discharge stroke = 0.2467 cubic foot at 605 deg. Fah. absolute temperature, and 14.8 lb. per square inch pressure.

From Table XIII., the exhaust gas was of 0.0820lb. per cubic foot average density.

: 0.2467 cubic foot weighed 0.2467 × 0.0820, = 0.02023 lb. at 492 deg. Fah. and 14.7 lb. pressure, =  $0.02023 \times \frac{492}{605} \times \frac{14.8}{14.7} = 0.01656$  lb.

at 605 deg. Fah. and 14'8 lb. pressure.



3. Air admitted per explosion occupied the total volume of the cylinder, less the volume of the exhaust products and gas. This was

0.8377 - (0.2467 + 0.0822) = 0.5088 cubic feet, and weighed

 $\frac{0.5088}{16.25} = 0.03131$  lb. per explosion.

TABLE XII.—SPECIFIC HEATS OF LONDON COAL GAS, USED IN TRIAL OF 7 HORSE POWER CROSSLEY ENGINE.

Constituent.	Per- centage by volume, per cent.	Weight Per cubic foot, lbs.	Weight of in 1 eubic foot of gas, lbs,	Percentage by weight, per cent.	Specific heat at constant volume, G v.	B.T.U.'s required to raise weight of constituent 1 deg. Fah. B.T.U.	Specific heat at constant pressure, Gp.	B.T.U.'s required to raise weight of constituent 1 deg. Fah B.T.U.
Marsh gas (CII,)	31.5	0.0447	0.01408	0.00 P	0-470	0-2011	0.593	0.2537
Olefines { (C, II, ) }	T-c	0.1174	66200-0	18-51	0.329	0.0654	101-0	0.0736
Hydrogen (H)	51.5	0.00359	0.00586	69.8	5.400	0.2091	3.400	5965-0
Carbon-monoxide (CO)	1-	0.0783	0.00003	18.33	0.173	0.0317	0.245	0.0449
Nitrogen (N)	3.0	0.0783	0.00235	17.14	0.173	0.0123	0-243	0.0178
Carbon-dioxide (CO <sub>2</sub> ) and oxygen (O)	1.8	0.123	0.00129	4.84	0.171	0.0083	0.516	0.0104
Totals	:	:	0.03290	100-00	:	0.5279	:	0-6961

TABLE XIII.—Specific Heats of Exhaust Products, for Trial of 7 Horse Power Crossley Engine.

Constituent.	Per- centage by volume, per cent.	Weight per cubic foot, lbs.	Weight of In 1 cubic foot of gas, Ibs.	Percentage by weight, per cent.	Specific leat at constant volume, Cv.	B.T.U.'s required to raise weight of constituent 1 deg. Fah. B.T.U.	Specific heat at constant pressure, C p.	B.T.U.'s required to raise weight of constituent 1 deg. Fab. B.T.U.
Carban-dioxide	91.9	0.123	0.0083	10.17	0.171	0.0174	0.216	0.0550
Oxygen	6.14	0.0802	0.0055	0.7.0	0.155	0.0104	712.0	0.0145
Nitrogen	87.10	0.0783	0.0682	83.13	0.173	0.1438	0.243	0.505.0
de commence de construction de				-				
Totals	:	:	0.0850	100.00		0.1716	:	0.2385

or

and

The mixture during compression therefore consisted of

0.001877 lb. gas; 0.01656 lb. exhaust products; 0.03131 lb. air.

Total = 0.049747 lb. per explosion.

The average specific heats of the mixture will be the specific heat of each constituent part multiplied by its relative weight. For the gas, the mean specific heats Cv and Cp must be calculated from the chemical analysis, as shown in Table XII., from which Cv = 0.5279, and Cp = 0.6961; or multiplying by 772, Kv = 407.5, and Kp = 537.4. Calculating similarly for the exhaust products, as shown in Table XIII.,

C 
$$v = 0.1716$$
, and C  $p = 0.2385$ , K  $v = 132.5$ , and K  $p = 184.1$ .

For air, the figures are  $Kv = 130^{\circ}20$ , and  $Kp = 183^{\circ}55$ . The mean specific heat of the mixture can therefore be found—as shown in Table XIV., page 88—to be,

K  $p = 199^{\circ}09$ , and K  $v = 141^{\circ}43$  foot-pounds,  $n = \text{K } p - \text{K } v = 57^{\circ}66$  foot-pounds, C p = 0.25788, C v = 0.1832,

Cv = 0.1832, $\gamma = \frac{Cp}{Cv} = 1.4077.$ 

Adopting these values, the temperatures will be-

$$\tau_b = \frac{67.8 \times 144 \times 0.2467}{57.66 \times 0.049747} = 840 \text{ deg. Fah. absolute};$$

$$\tau_c = \frac{840^\circ \times 240}{67.8} = 2973 \text{ deg. Fah. absolute};$$

$$\tau_d = \frac{2973 \times 0.2617}{0.2467} = 3154 \text{ deg. Fah. absolute};$$

$$\tau_\epsilon = \frac{48.71 \times 0.8377 \times 144}{57.66 \times 0.049747} = 2048 \text{ deg. Fah. absolute};$$

$$\tau_a = \frac{13.8 \times 144 \times 0.8377}{57.66 \times 0.049747} = 580 \text{ deg. Fah. absolute}.$$

It will be noticed that the temperature at A found by calculation (580 deg. Fah. absolute) is 25 deg. less than that (605 deg. Fah. absolute) previously assumed for calculating the relative weights of gas, air, and exhaust products in the mixture; but this difference does not affect the result, as all the three constituents will be increased in weight by the same amount, and their relative weights for the corrected temperature will be the same.

To draw the entropy diagram for the trial (see fig. 36), start with the mixture at B as the zero of entropy, when the entropy at C will be found from the formula, as explained previously—

$$\phi_{e} - \phi_{b} = C v \times \log_{e} \frac{\tau_{e}}{\tau_{b}}$$

$$= 0.1832 \times \log_{e} \frac{2973}{840}$$

$$= 0.23158.$$

$$\phi_{d} - \phi_{e} = C p \times \log_{e} \frac{\tau_{d}}{\tau_{e}}$$

$$= 0.25788 \times \log_{e} \frac{3154}{2973} = 0.01524$$

$$\phi_{e} - \phi_{d} = C v \times \frac{k - \gamma}{k - 1} \times \log_{e} \frac{\tau_{e}}{\tau_{d}}$$

$$= 0.1832 \times \frac{1.3707 - 1.4077}{0.3707} \times \log_{e} \frac{2048}{3154}$$

$$= 0.00790.$$

$$\phi_{a} - \phi_{e} = C v \log_{e} \frac{\tau_{e}}{\tau_{e}}$$

$$= 0.1832 \times \log_{e} \frac{580}{2048}$$

$$= -0.23112$$

$$\phi_{b} - \phi_{a} = C v \times \frac{k - \gamma}{k - 1} \times \log_{e} \frac{\tau_{b}}{\tau_{a}}$$

$$= 0.1832 \times \frac{1.3022 - 1.4077}{0.3022} \times \log_{e} \frac{840}{580}$$

$$= -0.02369.$$

### Balancing up the quantities of entropy, thus-

+		_
0.23158		
0.01524		
0.00790		
		0.23112
		0.02369
0.25472		0.25481
	0·23158 0·01524 0·00790	0·23158 0·01524 0·00790

The result shows a numerical error of about 0.04 per cent, which is practically negligible, probably due to slight inaccuracies of calculation.

TABLE XIV.—Specific Heats of Minture, for Trial of 7 Horse Power Crossley Engine.

Constituent.	Weight used per cycle, lbs.	K v for 1 lb., ftlbs.	K v for weight used, ftlbs.	K p for 1 lb., ftlbs.	K p for weight used, ftlbs.
Gas	0.001877	407.5	0.7649	537:4	1.0087
Exhaust products $\dots$	0.01656	132.5	2.1942	184.1	3.1487
Air	0.03131	130-2	4.0765	183.55	5.7469
Totals	0.049747		7*0356		9:9043

K 
$$r$$
 for mixture =  $\frac{7.0356}{0.049747}$  = 141.43 ft.·lbs. per lb.  
K  $p$  for mixture =  $\frac{9.9043}{0.049747}$  = 199.09 ft.·lbs. per lb.  
 $n = \text{K } p - \text{K } v = \frac{57.06}{57.06}$  ft.·lbs. per lb.

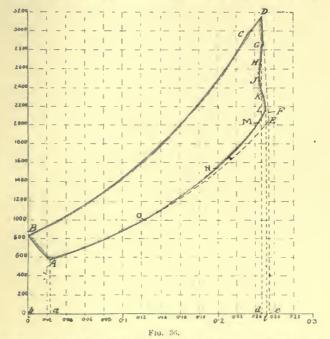
The  $\theta$   $\phi$  diagram for the ideal cycle is shown by ABCDE, in fig. 36, and if the area enclosed be measured by a planimeter, it will be found to equal 171.875 B.T.U. This represents the work which would be done by 1 lb. of gaseous mixture, and has to be corrected by multiplying by 0.049747

to get the work done by the known weight of mixture in the cylinder per explosion. Thus,

 $171.875 \times 0.049747 = 8.55$  B.T.U.

work done per explosion; or, expressed in work units,

 $8.55 \times 772 = 6600$  foot-pounds per explosion, which corresponds within 0.1 per cent with the value given



by Mr. Donkin (6,594 foot-pounds), as found by measuring the ideal cycle ABCDE of the pressure volume diagram shown in fig. 35.

27.—CORRECTED DIAGRAM FOR GAS ENGINE TRIAL

This ideal cycle has now to be corrected for the actual diagram, and, starting with the explosion period, the curves

BC and CD, in fig. 36, are correct for the actual diagram, but during the expansion period it will be seen that the actual pressure curve shown by the full line DGHJKL, fig. 35, differs very materially from the ideal expansion curve DE, shown dotted. It will therefore be necessary to take various additional points between D and F upon the actual pressure curve, such as G, H, J, K, and L, and calculate the temperatures and additional entropy at each of

TABLE XV.—Actual Expansion Curve: 7 Horse Power Gas Engine.

Position.	Pressure. Pounds per square inch.	Volume. Cubic feet.	Tempera- ture. Fah. absolute.	Index of expansion.	Increase of entropy. $d \phi$ .	Total entropy. $\phi$ .
			deg.			
D	240	0.2617	3154			0.24682
G	170	0.335	2858	1.3965	+ 0.000521	0.24734
н	134	0.394	0.250	1.4668	- 0.00175	
н	134	0.394	2650	1.4798	- 0.00185	0.24559
J	109	0.453	2478			0.24374
K	80.2	0.572	2312	1.2995	+ 0.00463	0.24837
				1.3526	+ 0.00190	
L	62.5	0.6897	2164			0.25027

these points. For example, at G the pressure is 170 lb. per square inch, and the volume 0.335 cubic feet; the temperature at this point will therefore be

$$\tau_g = \frac{170 \times 144 \times 0.335}{57.66 \times 0.049747} = 2858$$
 deg. Fah. absolute,

and the entropy at G above that at D will be

$$\phi_g - \phi_d = \mathbf{C} \, v \times \frac{k - \gamma}{k - 1} \times \log_e \frac{\tau_g}{\tau_d}$$

The value of k, the index of the expansion between D and G, will be

$$k = \frac{\log 240 - \log 170}{\log 0.335 - \log 0.2617} = 1.3965,$$

when  $\phi_g - \phi_d$  becomes equal to

$$0.1832 \times \frac{1.3965 - 1.4077}{0.3965} \times \log_e \frac{2858}{3154} = 0.000521.$$

The values of k, together with the temperatures, and entropy at the various points G, H, J, K, &c., are given in the accompanying Table XV., from which the actual expansion curve on the  $\theta \phi$  diagram, D G H J K, fig. 36, can be plotted and drawn.

For the exhaust period, the temperature and entropy can be calculated by the formulæ already given, as follows: At M, just after release, where the actual pressure shown by the indicator diagram is 53.8 lb. per square inch, and the pressure F on the ideal expansion curve at the same volume is 56.79 lb., the two temperatures will be

$$\tau_m = \frac{53.8 \times 144 \times 0.749}{57.66 \times 0.049747} = 2023 \text{ deg. Fah.,}$$
and
$$\tau_f = \frac{56.79 \times 144 \times 0.749}{57.66 \times 0.049747} = 2135 \text{ deg. Fah.}$$

The loss of entropy due to a drop in temperature from 2135 deg. to 2023 deg. at constant volume is equal to

$$d \phi = \phi_f - \phi_m = 0.1832 \times \log_e \frac{2135}{2023} = 0.00991$$

But this amount of entropy must be taken from that which the mixture would possess at F on the dotted curve. The entropy at F above that at D will be

$$\begin{array}{c} \mathrm{C}\,v \times \frac{k - \gamma}{k - 1} \times \log_{\epsilon} \frac{\tau_{f}}{\tau_{d}'}\,;\\ = 0.1832 \times \frac{1.3707 - 1.4077}{0.3707} \times \log_{\epsilon} \frac{2135}{3154}\,\mathrm{deg}.\\ \\ = 0.00713\\ \phi_{c} + \phi_{d} = \frac{0.24682}{0.25395}\\ \mathrm{then}\\ \mathrm{less} \qquad \qquad (\phi_{f} - \phi_{m}) = \frac{0.00991}{0.24404}\\ \end{array}$$

Similar calculations at other points in the exhaust curve of the indicator diagram, such as N and O, are made, as shown in Table XVI., when the actual temperature and entropy at O, coinciding with the theoretical curve E A, finishes the process.

TABLE XVI.—EXHAUST PERIOD: 7 HORSE POWER GAS ENGINE TRIAL

Pos	sition.	Pressure, lbs. abs.	Volume.	Actual Tempera- ture. Fah. abs.	Theoretical Tempera- ture. Fah. abs.	Entropy.
	M	53.8	0.749	deg. 2023	deg. 2135	0.25395 - 0.00991 = 0.24404
	N	38*0	0.808	1541	2076	0.255 - 0.055 = 0.200
	0	24.0	0.8377	1009	2048	0.2557 - 0.134 = 0.1217

#### 28.-Constant Volume Curves.

The two principal curves in the entropy diagram, fig. 36, viz., B C and E A, are seen to follow the general law of a logarithmic curve—

$$\phi_2 - \phi_1 = C \cdot \log_e \frac{\tau_2}{\tau_1},$$

and may be very easily drawn by the following geometrical method. Plot a temperature curve A B C D (fig. 37) to a base of its hyperbolic logarithms, on sectional paper, and between any two temperatures, say at B = 840 deg. Fah. absolute and C = 2,973 deg. Fah. absolute, draw the ordinates B M and C N on to any arbitrary base line X Y. From M draw the inclined line M O P, making an angle  $\alpha$  with the base X Y, such that tan  $\alpha$  = C, the constant in the above formulæ. Produce M O to intersect the higher temperature ordinate C N at P; then P N will represent to the same scale as the base the change of entropy  $\phi_2 - \phi_1$ , where  $\tau_2 = 2,973$  deg. and  $\tau_1 = 840$  deg.

If C be taken = C v for the trial just considered, viz., 0·1832, then the tan  $\alpha = 0\cdot1832$ , or  $\alpha = 10$  deg. 23 min., and P N will be found to scale 0·231. By drawing ordinates from the intersection of the curve with various temperatures between 840 deg. and 2,973 deg., such as 1,000 deg. Fah., 1,200 deg.

Fah., &c., the change of entropy from 840 deg. Fah. to these temperatures can be scaled direct from the line MP, being equal to the intersected portions of the ordinates at 1,000 deg. and 1,200 deg., or RS and TU respectively. By adopting this method of graphically finding the intermediate values of entropy for points between B and C, much time and calculation will be saved, and if the curve be drawn accurately and to an open scale, the result will be sufficiently

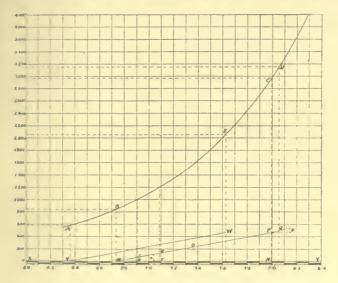


Fig. 37.

correct for all practical purposes. It should be noted that the most laborious part of the process, viz., drawing the logarithmic curve accurately, is only done once for all, and all other temperatures and values of C p and C v can be used on the same chart by drawing various inclined lines M P. The line V W, in fig. 37, is drawn for the period E to A of the 7 horse-power cycle just considered, the entropy at E being 0.23112 above that at A.

#### 29.—HEAT LOSSES IN 7 H.P. OTTO GAS ENGINE.

In the entropy diagram for the trial of this engine, shown in fig. 36, the heat usefully employed as work is represented by the area enclosed by ABCDJLO; but it does not represent the total heat evolved by the explosion of the gas. Knowing the weight of gas used per explosion (0.001877 lb.), and its calorific value (19,200 B.T.U. per lb.), the total available heat will be

$$0.001877 \times 19200 = 36.04 \text{ B.T.U.}$$

This must be represented on the entropy diagram, as in fig. 38, by producing the explosion line BC to P, so that the area b B P p shall be equal to

$$\frac{36.04}{0.049747}$$
 = 724.5 B.T.U. per lb. of mixture.

The theoretical temperature at P, due to complete combustion, can be calculated thus:

Let x = rise in temperature from B,

and  $C_v = \text{specific heat at constant volume}$ ;

then  $x \times C_v = 724.5$ ,

x = 3955 deg. Fah.,

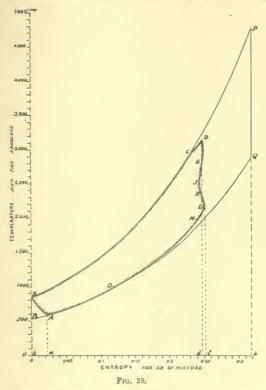
and temperature at

P = 3955 + 840 = 4795 deg. Fah. absolute.

The several losses can now be estimated direct from the entropy diagram, fig. 38, by measuring the areas they represent, as follows:—

The net work done = ABCDJLO = 8.20 B.T.U. per explosion, or 22.8 per cent of the total available heat. The heat given to the walls during compression, represented by the area a A B b, is equal to 0.77 BT.U. per explosion; that given to the exhaust gases a A O M L l = 13.63 B.T.U.; and the remainder, or l L J D C P p = 13.44 B.T.U., is transmitted through the cylinder walls. The total heat, therefore, given to the walls is 13.44 + 0.77 = 14.21 B.T.U., and this will be equal to the heat given to the jacket water, plus the radiation of the cylinder and piston. The former of these was measured during the trial, and found to be 14.02 B.T.U. per explosion,

thus leaving 0.19 B.T.U. for radiation. It is probable that some of the heat represented by  $\alpha$  A L l will have passed through the cylinder wall, and be included in that measured by the jacket water, so that the actual radiation will be



considerably in excess of 0.19 B.T.U. The details of the heat balance sheet, as measured from the entropy diagram, fig. 38, are given in Table XVII., and compare very favourably with the values given by Mr. Donkin in his report (Table V., Appendix A, of "Gas, Oil, and Air Engines").

With an ideal engine, assuming a non-conducting cylinder, complete combustion, and exhaust at constant volume, with adiabatic expansion and compression, the work done per explosion would have been that represented by the area

TABLE XVII.—HEAT BALANCE SHEET FOR 7 H.P. GAS-ENGINE TRIAL.

Area. (See fig. 27.)	Description.	B.T.U. per explosion.	Per eent.
b B C D d	Heat received during explosion	21.98	
dDGKLl	Heat received during expansion	0.62	• • • •
$b \to C \to L t$	Total received shown on diagram	22.60	62.7
$b \to P p$	Total received, calculated from weight of gas used	36.04	100.0
$\operatorname{L}\operatorname{D}\operatorname{C}\operatorname{P}p$	Difference = heat to jacket water	13.44	37.3
a A O $M$ L $l$	Heat to exhaust gases	13.63	37.8
$a \wedge B b$	Heat to walls during compression.	0.77	2.1
	Total heat lost	27.84	77.2
ABCDJLO	Heat in useful work done	8:20	22.8
<i>b</i> B P <i>p</i>	Total received	36 04	100.0

R B P Q, instead of A B C D L. The maximum work theoretically possible under these conditions is therefore equal to  $\frac{\tau_b - \tau_a}{\tau_b}$  of the total heat evolved, amounting in this case to

$$\frac{840 - 580}{840} = 0.3095 \text{ of } 36.04;$$

or 11:15 B.T.U. per explosion. The net work actually obtained in the cylinder, being 8:2 B.T.U., was only 73:5 per cent of this. The Carnot cycle of maximum theoretical efficiency, when applied to a gas engine, is therefore not only misleading, but incorrect, because the entropy diagram for a gas engine can never become a rectangle, as in the case of that for the steam engine.

#### CHAPTER VI .- APPLICATION TO OIL AND AIR ENGINES.

#### 30.—Entropy Diagram for 20 H.P. Diesel Oil Motor,

In general, the method of drawing the entropy diagram for an oil-engine test is similar to that employed for the gas engine: the temperature being calculated by the usual formula, P. V = R.T, and the entropy from the various equations used in the gas-engine trial, already described (see chapter v., page 75, &c.). As an illustration of a different cycle to the Otto, take that of the Diesel oil motor, described in The Practical Engineer, for May 6th, 1898. In this engine the compression stroke is followed by the admission of finely-sprayed oil injected by an air pump. and lasting from 5 to 10 per cent of the explosion stroke, according to the amount of power required. The ignition is effected by heating the air by adiabatic compression to a sufficiently high temperature as to cause the oil to explode. In other respects the cycle is similar to that of an ordinary gas engine. Taking the first trial made by Professor Schröter on a 20 horse power Diesel motor, the following particulars are taken from the Zeitschrift des Vereines Deutscher Ingénieure for July 24th, 1897 :-

Diameter of motor cylinderinche	s 9.856
Stroke of motor cylinderinche	s 15.725
Capacity of motor cylindercubic fee	et 0.695
Volume of clearance (assumed 62 per cent	t)
cubic fee	t 0.045
Revolutions per minute	171.8
Explosions per minute	85.9
Oil used per explosionlbs.	0.002122
Air used per explosionlbs.	0.039529
Total mixture per explosionlbs.	0:041651
Total mixture per expresson	0 041001

Professor Schröter estimates the mean specific heat of the gases at constant pressure at 0.264; and assuming the value of  $\gamma$  to be 1.408, the value of C v will be 0.1875; or, expressed

in foot-pound units, K p = 203.81, and K v = 144.75, and R = K p - Kv = 59.06 foot-pounds per pound of gas.

From this, knowing the weight of mixture present, and its pressure and specific volume, its temperature can be calculated in the usual way—

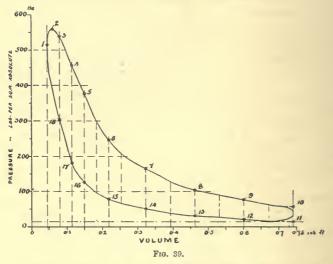
$$\tau = \frac{P \times V}{w \times R};$$

P = pressure in pounds per square foot (absolute);

V = specific volume of the mixture;

w =weight of mixture per explosion;

R = Kp - Kv for 1 lb. = 59.06 foot-pounds.



Thus, at point 1, at the beginning of the power stroke (see fig. 39), the temperature is—

$$\tau_1 = \frac{515 \times 144 \times 0.045}{0.039529 \times 59.06} = 1430$$
 deg. Fah.,

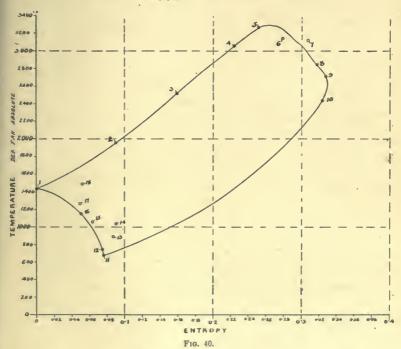
and at point 2, where the maximum pressure occurs,

$$au_2 = \frac{558 \times 144 \times 0.06}{0.041651 \times 59.06} = 1960$$
 deg. Fah.,

assuming the whole of the oil to have been injected during this portion of the stroke.

From 1 to 2 the increase of entropy can best be calculated by the following formula:

$$d\phi = \phi_2 - \phi_1 = C p \times \log_{\ell} \frac{V_2}{V_1} + C v \times \log_{\ell} \frac{P_2}{P_1}$$
$$= \left(0.264 \times \log_{\ell} \frac{0.06}{0.045}\right) + \left(0.1875 \times \log_{\ell} \frac{558}{515}\right);$$



that is to say, calculate the difference of entropy due to the increase of volume, as at constant pressure, and add to it the difference of entropy due to the increase of pressure, as at

constant volume. After point 2 this second quantity becomes negative, because the ratio of  $\frac{P_3}{P_2}$  is less than unity.

TABLE XVIII.—Entropy Diagram: 20 H.P. Diesel Motor.

Position.	Pressure. Lbs. per	Volume.	Absolute tempera-	Difference	of entropy.	Total entropy
Posi	square inch.	Cubic feet.	ture. Deg. Fah.	Positive.	Negative.	from position 1.
1	515	0.012	1430	0.09098		0
2	558	0.060	1960	0.06863		0.09008
3	539	0.0797	2516			0.15961
4	456	0.1145	3056	0.06413		0.22374
5	376	0.1492	3285	0.03380		0.25754
6	245	0.2187	3137	0.02062		0.27816
7	165	0.323	3120	0.02876		0.30692
8	105	0.462	2840	0.00974		0.31666
9	77	0.601	2709	0.01128		0.32794
10	56	0.740	2426		0.00478	0.32316
. 11	15	0.740	684		0.24699	0.07617
12	20	0.601	741		0.00038	0.07518
13	31	0.462	883	0.01273		0.08791
14	52	. 0.323	1036	0.00250		0.09041
15	78	0.2187	1052		0.02686	0.06355
16	125	0.1492	1150		0.01250	0.05105
17	180	0.1145	1271		0.00160	0.04945
18	305	0.0797	1500	0.00340		0.05285
1	515	0.045	1430		0.05285	0 00200
	010	0 040	11.00			
Totals				0.34657	0.34657	

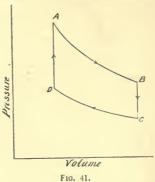
This formula has been used for calculating the entropy throughout the whole of the cycle, because the expansion and compression curves on the p. v diagram do not approach

sufficiently to a simple curve of the nature  $P_1 V_1^n = P_2 V_2^n$ . The mean value of n for the expansion period, from 15 per cent of the stroke to the end (points 5 to 10, fig. 39), is 1'1894; but its value varies very much during the expansion. The mean value of n for the compression period (from points 11 to 1) is 1.2629, but that also varies a good deal. The mean indicator diagram from the motor cylinder, shown in fig. 39, is reproduced from the translation of Professor Schröter's trial, as published in the Engineer of October 15th, 1897. All the figures for calculating the entropy are given in Table XVIII., and it is satisfactory to note that in this case the positive entropy exactly counterbalances the negative, a good proof of the accuracy of the calculations. The mean effective pressure of the diagram in fig. 39 is 1145 lb. per square inch, which equals 12,200 foot-pounds of work done per explosion. The area of the entropy diagram in fig. 40 is equivalent to 11,900 foot-pounds of work per explosion, or  $2\frac{1}{2}$  per cent less than that shown by the p, v diagram. This difference is probably due to the various assumptions made in calculating the specific heats. These calculations do not include the power absorbed by the air pump, which must be taken into account when calculating the net I..H.P., to obtain the consumption of oil.

#### 31.—Application to Stirling's Hot-air Engine.

Air engines may be divided generally into two classes—(a) those in which the air is heated at constant volume, and (b) those in which it is heated at constant pressure. Of the former class, Stirling's engine is perhaps the most common example; and the cycle of operations for such an engine is shown on the pressure volume diagram in fig. 41, and on the temperature fentropy diagram in fig. 42. In this engine the first operation is to admit a quantity of heated air at a temperature  $\tau_1$  from the regenerator to the motor cylinder, and expand it isothermally at the higher temperature  $\tau_1$  from A to B. The loss of heat due to the work done during expansion is

repaired by an external furnace, so as to keep its temperature constant during the stroke. At B the air is passed through the regenerator, where it deposits some of its heat; its temperature falling from  $\tau_1$  to  $\tau_2$ , and its pressure from B to C at constant volume. At C communication with the regenerator is closed, and the cooled air is compressed; isothermally at the lower temperature  $\tau_2$ , indicated by the curve CD; and, finally, it is passed through the regenerator again, to take up its deposited heat at constant volume; its temperature rising from  $\tau_2$  to  $\tau_1$ , and its pressure from D to A. The



thermal changes which take place will be better understood by reference to fig. 42, which shows the entropy diagram for the same cycle lettered similarly to fig. 41. Here the straight lines AB and CD represent the isothermal expansion and compression at  $\tau_1$  and  $\tau_2$  respectively; and BC and DA the constant volume curves, whose equation is:—

$$\phi = C v \times \log_{e} \frac{\tau_{1}}{\tau_{2}}$$

where  $\phi = \text{change of entropy between any two}$ temperatures  $\tau_1$  and  $\tau_{22}$ 

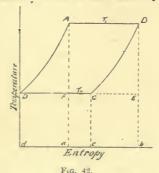
and Cv = specific heat of air at constant volume= 0.1686 B.T.U. per lb.

The heat supplied to the air during expansion is represented in fig. 42 by the area AB $b\alpha$ ; and that deposited by

the air in the regenerator by  $c\,C\,B\,b$ . That rejected during compression is shown by the area  $d\,D\,C\,c$ ; and that taken up in the regenerator by the air at the end of the cycle by the area  $d\,D\,A\,a$ . The net work done during one cycle is represented in heat units by the enclosed area  $A\,B\,C\,D$ , and the net heat supplied by the furnace (excluding the regenerator, which only acts as a reservoir of heat) being  $A\,B\,b\,a$ , the efficiency of the cycle will be the ratio

$$\frac{ABCD}{ABba}$$
.

But the two curves DA and CB are similar—that is, they are both logarithmic curves of the same equation between



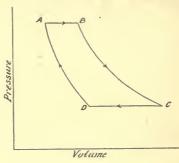
the same temperature limits, and therefore the area DAF will be equal to the area CBE. This must be so, because the amount of heat received from the regenerator is equal to the amount deposited in it, radiation being neglected. The area ABCD, representing the work done, will therefore be equal to the area of the rectangle FABE; and the efficiency becomes equal to

 $\frac{\text{FABE}}{\text{ABba}} = \frac{\tau_1 - \tau_2}{\tau_1}.$ 

Thus it can be geometrically proved without any mathematics that the Stirling hot-air engine has theoretically a maximum possible efficiency, being equal to that of the Carnot reversible cycle.

#### 32.—Application to Ericsson's Hot-air Engine.

The other class of air engines, viz., those that receive heat at constant pressure, can be similarly treated. For example, take the cycle of Ericsson's engine, shown with pressure and volume as ordinates in fig. 43, and temperature entropy ordinates in fig. 44. In this case, the first stage is admission from A to B of hot air at constant pressure, with its consequent reception of heat, represented in fig. 44 by a A B b. Then expansion isothermally at the higher temperature  $\tau_1$  from B to C, followed by the exhaust at constant pressure from c to D, the heat rejected being equal to



F10. 43.

 $c~{\rm C~D}~d$ ; and, finally, isothermal compression from D to A completes the cycle. The curves AB and CD, fig. 44, follow the equation

$$\phi = C p \times \log_{\ell} \frac{\tau_1}{\tau_2},$$

where Cp = specific heat of 1 lb. air at constant pressure = 0.2375.

The net work done per cycle and per pound of air is the area enclosed by ABCD, and the net heat supplied, exclusive of that received from and given to the regenerator, is b BCc; therefore the efficiency of the cycle is  $\frac{ABCD}{bBCc}$ . But the area aABb must be equal to the area dDCc; and the

work done, or ABCD, is therefore equal to EBCF; or the efficiency is

$$\frac{\mathbf{E}\,\mathbf{B}\,\mathbf{C}\,\mathbf{F}}{b\,\mathbf{B}\,\mathbf{C}\,c} = \frac{\tau_1 - \tau_2}{\tau_1}$$

as in the case of the Stirling engine.

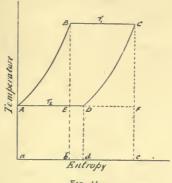


Fig. 44.

#### 33.—ENTROPY DIAGRAM FOR REFRIGERATORS.

The cycle of operations in refrigerators is exactly the reverse of that in the Carnot hot-air engine. Instead of taking in heat at a high temperature  $\tau_1$ , and transforming part of it into work, and rejecting the remainder at a lower temperature  $\tau_2$ , as in the heat engine, the working substance in the refrigerator receives its heat at the lower temperature  $\tau_2$ , and discharges it at a higher temperature  $\tau_1$ , the extra energy required being obtained from external work done on the gas. The theoretically perfect cycle that is reversible is shown in fig. 45 with pressure volume ordinates, and in fig. 46 with temperature entropy ordinates. The first stage of the cycle, A to B, consists of the adiabatic expansion of a certain quantity of air, the temperature falling from  $\tau_1$  to  $\tau_2$ . From B to C the expansion is continued isothermally at constant temperature  $\tau_2$ , the air receiving heat from the body which it is desired to cool, the amount of heat abstracted

being equal to the area EBCF, fig. 46. Compression commences at C, and is at first carried on adiabatically at constant entropy (or isentropically) from C to D, the

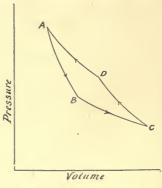


Fig. 45.

temperature rising from  $\tau_2$  to  $\tau_1$ , and is finally completed by isothermal compression from D to A, at constant temperature  $\tau_1$ , a quantity of heat being rejected to the water jacket equal to FDAE. The heat expended in the process is the equivalent of the work done on the gas, and is equal to the area ABCD in both diagrams. The heat absorbed from the substance to be cooled is equal to the rectangle EBCF, fig. 46, and the efficiency, therefore (in its thermo-dynamic sense), is equal to the ratio

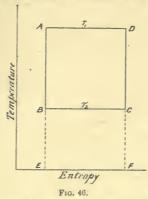
$$\frac{EBCF}{ABCD} = \frac{\tau_2}{\tau_1 - \tau_2}.$$

It is thus seen clearly how the efficiency is increased by reducing the difference of temperature between  $\tau_1$  and  $\tau_2$ ; and as the ratio

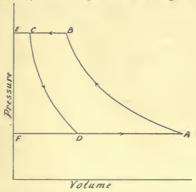
$$\frac{\tau_2}{\tau_1 - \tau_2}$$

may sometimes be greater than unity, it is better known as the "coefficient of performance" (see Howard Lectures, by Professor Ewing, on "The Mechanical Production of Cold," Society of Arts, 1897).

The series of operations in air refrigerators with an open cycle is somewhat different, and is shown in figs. 47 and 48. In this case the air is taken from the cold room and com-



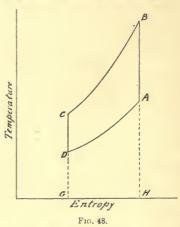
pressed adiabatically from A to B. It is then cooled at constant pressure, the temperature falling from B to C,



F10, 47,

fig. 48, and contracting in volume from B to C, fig. 47, after which it is passed into the expansion cylinder, where it expands adiabatically from C to D, and is discharged to the

cold room again. The work done on the air in the compression cylinder is equal to the area EBAF, fig. 47, or GCBH, fig. 48, and that done by the air in the expansion cylinder is equal to ECDF, fig. 47, or GDAH, fig. 48; so that the net external work required is the difference of these two quantities, represented by the area enclosed by ABCD in



both diagrams. The efficiency of the process will be represented by the ratio of the two areas,

$$\frac{ECDF}{EBAF}$$
, fig. 47;

but, as AB and CD are similar adiabatic curves, this will be equal to the ratio

$$\frac{E\,C}{E\,B}$$
 or  $\frac{F\,D}{F\,A}$ 

#### APPENDIX.

TABLE OF THE WEIGHT OF DRY SATURATED STEAM.

HAVING been engaged for some time past in carrying out and calculating the results of steam engine and boiler trials, the author has keenly felt the want of a more complete table of the properties of saturated steam than those hitherto published; one which would give the temperature, density, and total and latent heats for very slight intervals of pressure, in order to save the inconvenience of so much interpolation. In furtherance of this object, he compiled some time ago a new Table of the Density of Dry Saturated Steam, taking the well-known table of Prof. Dwelshauvers Derv. of Liege, as a basis, and extending it so as to give directly the density for every 10 lb. difference of pressure, with a list of additional weights which may be added to give the density for every 100 lb. difference in pressure; the latter, of course, only being used when extreme accuracy is required. It is published now for the first time, with the hope that it may be of some service to others engaged in similar work.

WEIGHT OF DRY SATURATED STEAM.

-							444										
		60	25		10		53		55		22			22			
	1p.	SO.	25		22		21		20		20			20			
	100	10.	20		19		18		18		17			17			
	nch re.	9	17		16,	-	16		15		15		-	14		_	
	for each	-05	14		77		13		13		12			12			
	ht f	.04	=		Ξ		10		10		10			10			
	eig	-03	00		00		00		00		1-			2-			
	Δ Weight for each 160 lb. pressure.	50.	9		9		rs		5		5			5			
		.01	63		03		co		63		61			2			-
		0-0	-00220	-00821	.01686	-01347	.01605	-01861	+1120.	.02366	-02316	·02:564	.03111	.03356	.03601	.03843	-04085
OTTOWN.	essure.	8-0	-00522	₹6200.	.01059	.01321	.01579	.01836	-02089	.02341	-02591	.02839	-03087	03333	.03576	03819	.01061
C C C	no lb. pr	2.0	100484	70700-	.01033	-01295	-01554	-01811	-02064	.02316	.02566	-02814	-03062	.03307	-03552	-03795	.04037
TWIT TWO	for each	9.0	-00467	00140	-01007	.01269	.01528	.01785	-02039	.02291	-02541	.02789	-03038	-03283	-03527	-03770	.04013
7 777	ibic foot	0.2	-00439	.00713	08600.	.01243	.01502	.01760	-02013	.02265	.05216	.02764	.03013	.03258	-03203	-08746	68680.
	ds per cu	1.0	.00411	98900.	-00953	-01217	01476	.01734	.01938	.02240	-02491	-02740	.02988	-03234	.03478	.03722	-03962
1	Weight in pounds per cubic foot for each $\gamma_0$ lb. pressure.	0.3	-00383	-00658	12600-	06110-	.01450	.0110	.01963	-02215	.02466	.03715	-02964	-03200	-03454	-03698	-03910
	Weigh	ē-0	-00355	-00631	00600-	-01164	.01424	.01683	.01938	-05190	-02441	06960.	-07039	-03185	-03430	.03673	.03916
		0.1	-00327	+0900-	-00874	.01138	.01399	76910.	.01912	.02165	-02416	.02666	-05914	.03160	-03405	.03649	-03892
Andrew Company	Δ Weight	per Ilb. press.	.00278	.00271	-00264	-00201	-00528	.00256	.00258	-00251	.00220	-00548	-00247	-00245	100544	-00243	-00242
and the same of the same of	> 7	cubic foot,	-00599	-00277	-00848	-01112	.01373	-01631	.01887	.05140	.05301	-02641	68850-	.03136	18880.	-03625	-03868
- 4	sure, per l, abs.	l.bs.	-	5	63	4	10	9	-1	00	6	10	11	12	13	14	15

4	Th	T	323	3.7	D	Ŧ.	1	ge .
a.	R.	B.	Etc.	20	17	т.	-3	٠.

			A	PPE	ND	IX.							1
67			54						21				1
2			10						10				
1-			16						16				-
NP pret		-	***						Pre				1
15	-		12						12			_	
10			10						0			_	1
1-			1-						1-				1
10			10						ю				1
61			01						63				
.04567	-05046	-05521	.05758 -05994	-06239	.06463	10990-	-06931	.07161	-07397	02639	-07860	-08092	
-04303 -04543	.05022	.05497	-05970	-06206	.06410	81990.	-06908	-07141	-07373	-07600	-07837	69080-	
.04279 -04519	.04998	.05473	.05917	.06182	.06416	-06650	.00384	•07118	-07350	68220	-07814	-08046	
.04255 .04495	.04974	-02420	05687	.06159	-06393	.06627	.06861	16010	.07327	02220.	16270	.08023	
04230	-04950	-05426	-05663	-06135	.06370	-06603	-06838	.07071	.07304	-07536	-07768	00080-	
.04147 .04487	-01926	-05403	-05640	06112	.06346	.06580	-06814	-07048	.07281	•07518	-07745	77670-	
04182	-04902	-05879	05816	-06088	.06323	106557	16299.	-07025	-07258	-07490	.07722	107954	
.04158 -04399 -04639	01810-	-05355	-05592	-06065	-06299	-06533	-06768	.07001	-07234	-07466	86920.	-07931	
04134	-01855	-05332	-05569	-06041	.06276	01290-	-0674-4	84690.	.07211	-07443	-07675	-07907	-
.00240	-00239	-00237	-00236	-00235	-00-34	-00234	.00234	.00233	-00232	-00232	-00232	18200-	
-04110 -04851 -04591	-04831	-05308	-05545	.06018	.06253	-06187	.06721	-06925	-07188	.07430	-07652	07884	-
16	13	5	el 51	G-5	25	26	101	808	61	30	31	97	

Continued on next paye.

WEIGHT OF DRY SATURATED STEAM-continued.

		0.0	1	2												07		
	P.	50.	1	oc												20		
	100	10		91												16		
	re.	90.		14												7		
	t for each	0.0		Ξ						_						1		
	it fo	10	İ	6.												Œ.		
	eigl	.03 .04 .05		1-												t-		
	Δ Weight for each 13a lb. jacssure.			10						-						23		
	1	.01 0-5	i	Ç1												©1		
			93	20	63	61	=	0	00		9	7	-	00	53	0	9	01
naed		6.0	.08323	.08353	-05783	-09012	.09241	.09470	86960-		95660.	10154	10381	10608	10835	11060	.11286	11512
-conti	essure.	8.0	.08300	.08230	09280-	68680	-09218	-09447	-09675		.00600	10131	10358	10586	10812	11037	11263	-11490
TEAM	η Ib. pr	1-0	.08277	10650-	·08737	99680.	.09195	£6\$60.	25960-	1	0.0881	10108	10336	10563	.10789	11015	11911	11467
ATED A	for each	9.0	.0S254	.08484	.08714	.08943	.09173	.06401	65960-		.08228	10085	10313	10540	10701	10992	11918	11445
SATUR	thic foot	0.2	.08231	-08461	108691	-08920	09160.	-09378	90960-		0:1830	10062	10290	10518	10744	0.1001.	11196	11422
DEY	ds per cu	\$ 0	.08508	.08438	89980.	16880.	.09127	.09355	.09583	0	21860.	10040	10268	10495	10721	1601.	11173	11400
WEIGHT OF DRY DATURATED STEAM - continued.	Weight in pounds per cubic foot for each $1_d$ lb. pressure.	0.3	-0S185	.08415	.08645	.08875	.00104	.09332	.09260	0.000	.0.7.30.	10017	10245	10472	.10699	10925	111151	11377
W Elti	Weight	6.5	.08162	568SO.	-08622	-08852	18030-	.09310	-00538	HO MOOD	0000	£6660.	10555	6++01.	10676	10005	-1112s	11355
		0.1	.08138	.08369	.08299	65880.	.0905S	-09287	-09515	9 8 200	0.00	-09972	10200	.10457	10654	.10880	.11106	11332
	A Weight	per 1 lb.	.00231	.00530	-00230	.00256	00550	66600	-00228	00000	00223	-00558	12200.	.00227	-00226	-002243	-00-256	-00226
	weight.	enbic foot.	.08115	-08346	.08226	90880-	-09035	1956).	-09493	10400.	087.21	-09949	10177	10404	10631	10821.	.11083	.11300
	sure, per 1, abs,	8d,	90	37	25.5	36	17	38	39	9	0.5	41	62	43	44	45	46	4-

1	
	- 0
	0
	200
	-
	- 290
	~
1	- 52
1	-
1	
	200
	10
	239
	0
	(10)
	- (%)
	Cop
	- 22
1	0
	_
	-
	~
	-
	- 29
	-
1	~
	. 5
	~
Į.	~
	~
	ಿ
9	13

					50											
					18		-					 				
					16											
					13											
					Ξ											
					6											
					1-											
					4											
					61											
11787	11962	12187	12412	12636	12860	13083	13306	18528	13751	13975	14197	11419	14641	.14863	15085	15307
11715	.11940	12165	12390	12613	12888	13061	13284	13506	18729	13952	14175	14397	.14619	.14841	.15063	15285
11692	11911	12142	12367	12591	12815	13039	13261	13483	10181.	13920	14158	.14875	14596	.14818	15041	15263
11670	11895	12120	12845	12568	12793	13017	18289	13461	13684	13908	.14130	.14852	14574	14796	15018	15240
11647	27811.	12097	12322	12546	12770	12994	13216	13439	13662	13885	.14108	.14330	.14552	14774	.14996	15218
11625	11850	12075	12300	12523	12748	12972	13194	13417	13640	13863	14086	.14308	.14530	-14752	14974	15196
11602	11827	12052	12277	12501	12725	12949	13172	13395	13618	13841	.14064	.14286	14508	14730	14952	15174
11580	11805	.12030	12255	12479	12708	12921	13149	13872	13595	13818	14041	.14263	.14486	.14708	14980	15152
11557	11782	12007	12232	12456	.12680	12904	13151.	13350	13573	13796	14019	-14241	-14463	.14685	14907	15129
-00225	200225	-00225	-00224	-00554	-00554	.00224	.00223	-00523	-00553	-00-53	-005555	.00555	-00555	-00222	-00555	-00222
-11585	11760	11985	12210	12434	.12658	12882	13105	13328	13551	13,74	13897	14519	14441	.14663	14885	15107
8	67	3	51	52	58	54	22	26	150	28	59	3	61	62	63	64

WEIGHT OF DRY SATURATED STEAM-continued.

	1	0.0	20										20					
	1b.	80	18										18					
	100	10.	16										15					_
	ach re.	90.	133										63					-
	t for eacl pressure.	.05	11										11					_
	pre pre	†O.	6										0					_
	cigl	<u>م</u>	1-										1-					_
	Δ Weight for each 14σ lb. pressure.	-05	4										491					
		10	2										C3					
enen.		6.0	.15529	15750	15970	16190	.16410	.16630	16849	17068	17287	17506	17725	17944	18163	18382	18601	
-continued.	eseure.	8.0 -	15567	.15728	.15948	16168	.16388	.16608	16827	17046	17265	17484	17703	17922	18141	18360	18579	
OLEAN W	Weight in pounds per cubic foot for each 1,6 lb, pressure.	2.0	15485	15706	15926	16146	16366	16586	16805	17021	-17243	17462	17681	17930	18119	18338	18557	
	for each	9.0	15462	15684	15904	16124	16344	16564	16783	17002	17221	17440	17659	17578	76081.	18316	18535	
Calonalan	ibic foot	0.2	15440	15662	15882	16102	16322	16542	16762	18691	17200	17418	17637	17856	18075	18294	.18513	
LAIL L	ds per ct	1.0	15418	15640	15860	16080	16300	16520	16740	16959	17178	17397	17615	17.834	18053	18272	18491	
II OL	mod uj	0.3	15396	15617	.15838	16058	16278	19498	16718	16937	17156	17375	17594	17813	18031	18250	18469	
II DIGILL	Weight	0.5	15374	15595	15816	16036	16256	16476	16596	16915	17134	17353	17572	16771.	18010	18230	18148	•
		0.1	15351	15573	15794	16014	16234	16454	16674	-16893	17112	17331	17550	17769	17988	18207	18426	
	Weight	per 11b. press.	•00252	.00221	.00220	.00550	05500.	-60220	.00219	•00219	01200-	.00219	.00219	.00219	-00219	01200.	-00510	
į	Weight.	cubic foot.	15329	15551	15772	15992	16212	16432	16652	16871	06011.	17309	17528	177.47	17986	18185	18404	
	sure, per per per	Pres.	92	99	19	89	69	0.1	17	12	13	-1 -1	12.	1.6	11.	27	1.0	

									24.1	LEW	D.	7.7.								1
					30										10					,
-	-	-		_	-1								_	_	1-		-			1
-					15										15					
-					002 []										13					-
-					11										Ξ					-
					0										6					-
-					1-										9					-
-					4										4					
					2										2					
	18819	18061	19255	19473	16951	.19909	20127	-20343	-30260	11105.		-50605-	-21211	-21427	21643	-21858	-22073	-55588	-22503	and the same of the same of
•	86231.	19015	.19233	19421	69961.	19881.	-20102	-20322	-20539	-20756		2002-	.21180	-21402	21621	-21837	-22052	19555	-22482	- manual
	18776	18994	119211	19429	19647	19865	-20083	-20300	20217	-20734		-2000	-21167	-21383	.21299	-21815	-22030	-99945	-22160	-
٠	18754	18972	06161	19407	19625	19843	-20061	-20278	-50495	20712		.20959	-21146	21362	81212	-21793	60075-	-25554	-92139	
	18732	18950	19168	19386	10001	19822	.50040	20257	£250g.	.20691		-20002-	.21124	-21840	.21556	21112	-21987	-55505	-22417	
	18710	87081.	19146	19364	10582	19800	20018	-20235	20452	69905.		98803	.21103	-21319	21535	*21750	-21966	-22181	-22396	
	18688	18906	19124	19342	19560	81161.	19996	20213	-20430	-50647		-20864	.21081	70212	.21513	-217.29	-21944	-22159	-22374	
	18666	18385	19103	19320	10538	19756	19974	20102	-204.8	-20625		.20842	21059	-21275	*21491	-21707	-21923	.22138	-22353	
	18645	.18863	19061	19200	11961.	19734	19952	-20170	20387	+0907-		.20831	-21038	-21254	.21470	-21686	-21901	.22116	-2-2331	
	-00218	-00-13	\$1200.	-60218	200-218	-00218	-00217	-00217	-00217	-00517		-00217	00210	00216	00210	-00216	-00215	-00215	.00215	
	18628	18841	19059	11561.	19495	19718	19931	81107.	-20365	28202.		-20280	91012.	-91982	-21448	-21664	.21860	25095	.22310	
	os So	SI	33	88	84	2	98	29	8	80		8.	91	92	93	₹.	95	96	16	1

Continued on next paye.

WEIGHT OF DRY SATURATED STEAM—continued.

		60.							19							
	1p.	-08							17							
	100	10.	1						15							
	ach	90.	1						13							
	t for each pressure.	-05							Ξ							
į	ht f	ţ							6							
	eig	.03							9							
	A Weight for each 100 lb. pressure.	01 02 03 04 05 06							+							
		.01							63							
retector.		6.0	-22718	-22933	-23148	-23363	-93578	-23793	.24008	.24223	.24138	.24652	-21866	-25079	-25292	-25506
DALOMATED NIMAM CONCERNIC	essure.	8.0	-22697	-22912	-23127	-23342	-23557	-23772	-23987	.24505	.24417	-24631	.24844	-25057	.25271	-25485
THE WITTER	le lb. pr	7.0	-22675	-22890	-23105	-23320	-23535	-23750	-23965	-24180	-24395	-24610	.24823	-25086	-25249	-25464
THE TOTAL	for each	9.0	-22654	.22869	-23034	-23299	-23514	-23729	.23944	.24159	-24874	-24588	-24802	-25015	.25228	-25442
TOTO	ibic foot	9.0	-22632	.22847	-23062	-23277	-23492	-23707	-53055	-24137	-24352	.24567	-24780	-24993	.25207	-25421
T T T	ds per cu	<b>7.</b> 0	-22611	-22826	-23041	-23256	-23471	-23686	10082.	.24116	-24331	.24546	.24759	-24972	-25185	.25399
HEIGHT OF	Weight in pounds per cubic foot for each $1^{\rm b}_{\rm l}$ lb. pressure	0.3	-22589	-22804	-23019	-23234	-23440	-23664	-23879	-5400+	-24309	-24554	.24738	-24951	-25164	-25378
1	Weight	~0	-22568	.22783	-22098	-23213	•23428	-23643	-23858	.24073	.24288	.24503	.24716	.24929	-25143	-25357
		0.1	-22546	-22761	-22976	-23191	-23406	.23621	.23836	.24051	-24266	-24481	-24695	.24908	-25121	-25835
	Δ Weight	per I lb. press.	-00215	.00215	.00215	-03515	·00215	.00215	00215	-00215	.00215	-00214	-00213	-00213	-00214	•00214
	weight.	cubic foot.	-22525	-22740	-22955	-23170	-23335	-23600	-23815	.24030	-24245	-24460	-24674	-24887	-25100	•2531
	per per share,	841	86	<u>6</u>	100	101	102	103	104	105	106	107	108	109	110	ш

									A	PPI	ENI	XIO								1
	13			19								19								-
	-1			17								17								-
	15			15								15								-
	53			13								13								
	=			=======================================								1								
	0			6								00								
	9			9								9								
	44			4								-44								-
:	61	-	-	63	_	_						61								
	-25720	-25934	-26148	-26363	-26576	-26790	-27004	-27217		-27430	-27643	-27854	-28066	-28278	-28490	-28702	-28914	-29126	-29338	
	.25699	-25913	.26127	.26342	-26555	-26769	.56983	27186		-27408	.27621	-21883	28045	.28256	.58469	.286SI	.28893	-29102	-29317	
•	-25678	-25892	-26106	-26321	-26534	-26748	.26961	-27174		1381	.27600	.27812	-28024	-25232	28448	09985.	2882.	\$8067-	96565-	-
	.25656	25870	.26084	66595.	-26512	.26726	01695.	-27153		-27366	.27579	-27791	28002	-58514	-28426	-28638	.28820	-29062	-29275	
	-25635	.25840	-26063	-26278	-26491	-26705	.26919	-27132		-51345	-27557	-27769	18642-	-28193	-28405	-28617	.28829	-29041	-29253	
	-25613	-25827	.26041	-26257	-26469	-26683	.56891	-27110		-27323	.27536	.27748	09615-	-28172	·28384	-28596	.28808	.29020	-29232	
	-255592	-25806	.26030	-26235	.26448	-26662	26876	05015.		-27802	-27515	-27727	.27.939	-28151	.28363	-28575	18181	66685.	-29211	
•	-25571	25785	-25999	-26213	-26427	-26641	26835	-27068		-27281	-27494	.27705	71072	.28130	-28841	-28554	.28766	-28978	-29100	
	-25519	-25763	.25977	26195	-26405	-26619	-26833	-27046		-57259	-27472	.27684	93815-	-28108	-28320	-28532	-28744	92685-	-29168	
	.00514	-00214	.00214	-00214	-00214	1000	.00213	1500.		.00213	.00512	.00212	-00212	-00212	.00212	.00212	-00-112	.00212	-00212	
	-2552S	-25742	-25956	-26170	-26384	-26598	-20812	27025		-27238	-27451	-27663	-27875	28087	·28299	.28511	.28723	\$8032	-29147	
	112	113	114	115	116	117	118	119		120	121	122	123	124	125	120	127	128	129	

Continued on next page.

WEIGHT OF DRY SATURATED STEAM-continued.

	1	60.	1			13											
	.e.	30.	i			4 T											
	100	19.	i			15											
	nch re.	90.	!			13											
	A Weight for each 186 lb. pressure.		1			-											
	pr fe	40				00											
	elgl	-02 -03 -04 -05				9											
	7	-0.	1			4											
		0.				63											
eucte.		6-0	-29550	29762	\$ 166Z.	93106-	.30398	30000	-80830	.31031	.31242	-31453	-31664	31875	·32086	-32297	-32508
CALCINETY DIEAM -CONCINGED.	essure,	8-0	-29529	-29741	-29953	30165	.30376	.30588	-80799	.31010	-31221	-31432	-31643	.31854	-82065	.32275	-32486
THE PARTY.	Weight in pounds per cubic foot for each $_{1^{\prime}\! 0}  \mathrm{lb.}$ pressure.	1-0	80265-	-29720	-29933	.30143	*30355	-30567	.30,78	.30989	.31200	.31411	-31622	-81833	.35043	-32254	-82465
A LELL	for each	9.0	-29487	-29698	-29910	-30122	-30334	-30546	30757	-30968	-31179	.31390	.31609	-31812	.32022	-85538	.32414
777	bic foot	0.2	-29465	11967-	68865-	.30101	-80313	-30525	-30736	-30947	-31157	.31368	-31579	.31790	.32001	-89912	-32423
17161	ds per cu	1.0	-20144	-29626	89867	.20080	-30292	.30203	*80714	-30925	.31136	-31847	-31558	-81769	-81980	-32191	-32102
THE THE LAND IN THE	in poun	0.3	-29423	-29635	-29847	-30029	-30270	-804 12	.30693	.30904	31115	.31326	.31537	31748	-31959	.32170	·32381
200	Weight	0.5	-29402	-59614	29856	-80037	-30549	-80461	30672	-30883	31091	-31305	31516	-31727	-31938	.55140	.32350
		0.1	.29380	-29592	10867	-30016	.30228	.30140	-80651	-30862	.31073	31284	-31495	31706	-31917	.32128	.35333
	A Weight	per I lb. press.	-00212	.00212	-00212	-00-12	-00212	.00211	.00211	-00211	.00211	.00211	-00211	-00211	.00211	-00211	-00211
	Weight.	cubic	-29829	-29571	-297.83	36062	-80207	.30419	.80630	.30841	.31052	.31263	-31474	.31685	-81896	.82107	-32318
	per per abs.	add ii ,pa	180	131	132	133	184	185	136	137	188	139	140	141	143	143	111

								Al	PPE	ND	IX.							
	19							10										
	17							1-										
	15							15										
	13				_			33										
	10		_					10								_		
	90			_				90										
	9	-	_					9										
	4																	
	- 63							61								_		
																_		
-32718	-32928	.93138	-33348	.33558		.33768	11688.	.34186	.34305	34004	.34813	-35622	*35231	.35440	.35649	٠	-35858	29098.
70528-	32907	.33117	.33327	-33537		7. 185. 7. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1.	*33957	34165	.34374	232F2.	.347.92	.35001	.35210	.82419	.35628		.25837	.36046
.32676	.82886	.33696	-33306	.38516		.33726	.33936	.34145	.34353	.34562	.34771	-34980	.35189	35393	.35607		.35816	36025
-32655	32865	.33075	38288	33-195		20183.	-33915	.34154	-34333	.34543	.34750	-34959	.35168	35377	-35586		-85795	-36004
-32534	.32844	-33054	-33264	-33474		.33684	.33894	.34103	.34312	-31521	-34729	24933	-35147	35356	35565		35774	-35983
.82613	-85828	-33033	-33543	-33453		.38663	.33873	34085	-34291	.34200	-31709	34918	35126	-37.333	35541		.35753	-35962
-32592	.32802	-33012	-83922	.83485		-33642	-33852	.34061	.34270	34479	.34688	76816.	.32100	-35315	-35523		.95739	-35941
-32571	.32781	-82991	.33201	-33411		.33621	-33831	-34040	31219	-34458	-34667	.34570	.32085	-35294	-33503		-35712	.35920
-32550	.32760	.32970	-33180	.33390		.33000	.33810	.34019	36616.	3443	.34646	-3 1855	.35064	.9:273	-35482		.35691	-35000
01200-	-00-10	-00210	-00210	-00210		-00310	.00209	.00500	-00200	-00500	-00200	-00200	60500-	-00500	-00200		60,00.	60500-
.82529	-82789	85558	.33159	-33369		.33579	-33789	-33008	.34207	.84416	-34025	-31834	-32043	-85252	-35461		.35670	.35879
145	146	147	148	149		150	151	152	153	154	155	156	157	158	159		160	191

Continued on next page.

Weight of Dry Saturated Steam-continued.

		8		19												13		
	1b.	-0.8		1,												11		
	100	10.		15												15		
	re.	90.		12												12		
	t for each	-02		10												10		
	A Weight for each 140 lb. pressure.	\$		00			_									00		
	eigh	-03		9							_				_	9		
	I W	-05		4												-4-		
	7	.01		61								_				67		
			10	62	_	0	1-	10	82			6	1-	10	63	-	6	l n
nana.		6.0	-36275	.36483	.36691	.36899	.37107	.37315	*37523	.37731		.37939	-38147	-38355	.38563	11188.	.38979	-39187
-coner	ssure.	8.0	.36254	.36462	.36670	.38878	.37086	*37294	-37502	.87710		81618	.88126	.38334	.88242	.38750	.38958	.39166
DIEAM—COntinued	70 lb. pro	1.0	.36234	.36442	.36650	.36857	-37066	-37274	-37482	.37690		81898	.38105	.38314	-38522	.38730	.38938	39146
	for each	9.0	-36213	.36421	.36629	.36837	.37045	-37253	.37.461	.37669		11818.	-33082	.38293	.38501	.38709	.38917	89125
SALUKALED	bic foot	0.2	36195	-36400	3660s	.36816	-37024	.37232	-37440	.37648		.37856	-38064	.38272	-38180	.38688	38896	39104
DKY K	Weight in pounds per cubic foot for each 15 lb. pressure.	0.4	.38171	.36379	.36587	36795	.37003	.37211	.37419	.37627		-37835	.38043	.38251	-38459	.38667	.3887.5	-39083
HI OF	in poun	0.3	.36150	.36358	.36566	36774	.36982	.37190	.37398	.37606		-87814	33022	.38280	-38438	.38046	-38854	-89062
WEIGHT	Weight	0-5	.36129	.36338	.36546	-36754	36942	.37170	.37378	.37.556		-87794	38002	.88210	.88418	.38626	.38834	-39012
		0.1	-36108	.36317	.36525	.86783	.36941	.37149	-87357	-37565		-37773	.87981	.38189	-38397	.88605	.38813	-39021
	A Weight		.00208	-00208	-00208	-00200	-00203	-00208	-002008	-00208		-00508	80500-	.00208	-00508	-00508	-00208	.00208
		cubic foot.	36038	-36296	.36504	-36712	-36920	.37128	-37336	-37544		-37752	.37960	-38168	.88376	-38584	.387.92	-39000
	per abs.	mr the l	162	163	164	165	166	167	168	169		170	171	17.2	173	174	175	176

								10										
			_											_				
								60										
								21										
								10			-							
								00										
								9										
								4										
								C1										
-39895	.39003	-30811		.40010	.40227	40435	.40643	.40821	.41059	.41267	.41475	.41683	.41891		.45000	15307	44515	.42723
-39374	.89989	-39790		.39998	.40500	.40414	.40622	.40830	.41038	.41246	*41454	.41662	.41870		.45078	.42280	16565.	20124-
-39354	-39562	.39770		-89978	93104.	.40394	.40602	.40810	*41018	.41226	•41434	.41642	·11850		.42028	*42266	-12474	.42682
-30838	-39541	-39749		-39957	.40165	.40873	.40581	.40789	40997	.41205	-41413	.41621	41829		-42037	.42245	.42453	.42061
39312	-39520	.39728		.39936	.40144	.40352	.40200	\$9201.	97604.	41184	-41395	.41600	.41808		.42016	.42224	.45435	.42640
199991	-39499	.39707		-39915	.40123	10331	.40539	-40141	40955	.41163	41371	41579	.41787		41895	.42203	.42411	.42619
02768.	-39478	39686		-39S94	-40105	.40310	.40518	.40726	.40034	.41142	.41350	.41558	41766		+1974	.42182	.45300	.42598
.39250	-39458	-39666		-39874	70085	.40:50	\$640£	.40,06	*40914	41122	.41330	-41538	911140		+11954	.45162	.42370	.42578
-39229	.39137	-39645		.30823	-10001	40209	10477	40083	.40803	.41101	41300	41517	.41725		.41988	.42141	.45349	.42551
807-00-	-00508	-00208		.00500	-00503	-00500	.00208	807.00-	80500.	-00508	-002008	-00500	-00508		80500-	-00508	.00500	-00208
39208	-39416	-39624		.30832	.40040	.10248	.40426	10664	5280t.	08017	.41288	.41496	+11704		-41912	.42120	45828	.45236
177	100	17.0		180	181	182	183	184	185	186	187	188	180		100	191	19-2	163

Continued on next page.

WEIGHT. OF DRY SATURATED STEAM-continued.

1	. 1	60	19							10				10				
	۵.	-08	17						 	1-				2				
	Ton	10.	15						 	271			_	40				
	reh	90	12							21				67				
	Δ Weight for each 14π lb. pressure.	.05	10						 	10				10				
	t fo	70	00						 	90				00				
	eigl	.03 -04	9						 	9				9			_	
u	1	-0-2	4	-					 	-Pr				4			-	
	7	.01	C1							23				01			_	
e eccono		6-0	.42931	43139	.43347	*43555	-43763	-43971	.44178	.44386	.44592	-44799	.45006	-45212	.45418	-45623		
Contracta	ssare.	8.0	.42910	.43118	.43326	+4353+	43749	.48020	.44158	.44365	144571	.44778	.41985	.45191	-45397	*45602		
TI ELLIN	lo lb. pre	2-0	.42890	860gF.	-43300	.43514	12784	.43929	.44137	11811	.41550	89215.	14964	.45170	-15376	*455582		
DALOWALED DIEAM	Weight in pounds per cubic foot for each $^{10}_{10}\mathrm{lb}.$ pressure.	9.0	42869	13077	*43285	.43493	.43701	.43908	-44116	.14323	.44530	.44737	11011	.45150	45356	-45561		
		0.0	.42848	.43026	*43564	-13415	.43680	-43888	.44095	.44303	.44510	.14716	.14928	.45129	.45335	.45541		
LANCE	ds per cu	0.4	12821	.43035	.13243	.43451	.43659	13867	-14075	-44282	.44189	96911.	.44903	.42109	.45315	.45510		
WEIGHT OF	in poun	6.0	.42806	.43014	-43000	.43430	.43638	.43816	14021	.44261	.41468	.44675	*41882	.45088	-45294	.45500	_	
W EIG	Weight	0.3	.42783	45994	-43202	.43410	.13617	.43855	.44033	.44240	.41418	14654	.44861	19061	.45273	.45479		
		0.1	.42765	-42973	-43181	.43389	*43597	43805	.44013	.44550	.41427	.44631	-44841	74064	.45253	.45459		
	A Weight		.00200	•00208	*00200	-00508	-00200	-00208	10500-	-00307	10200-	10200.	-00200	00500	-00200	.002002		
	Weight.	cubic foot.	-42741	.45935	.43160	.43368	.43576	.43784	-43965	.44199	.41406	.44613	.44820	.45026	.45232	-45438		
	per per	Press. I.bs.	194	195	196	197	198	199	200	201	203	203	201	202	206	207		

											-			
		18						00						
-		9						ř						
_		14						7						
-		12						12						
		10						10						
		90						00						
		9						g						
		791						-de						
-		C/I	_					61						
	*45828	.46032		.46237	.46441.	16614	.46848	-47052	47255	-47457	-47659	.47861	.48063	
	-45807	.46012		.46216	46420	.46624	.46828	-47032	-47234	-47437	.47639	11.841	-45043	
	45787	-42883		96194	46400	.46604	10804	.47011	-47214	.47416.	.47618	.47.820	.48022	
-	-45766	.45971		.46175	.46379	.46583	-46787	.46991	£612£.	.47396	.47598	47800	-4800 <sub>2</sub>	
_	.45746	.45951		.46155	.46329	.46563	.46766	14691	+1114	-47376	-47578	03216.	6861F.	
	45725	.45930		.46135	.16339	*46548	.46746	.46950	-47153	.47356	-47558	.47760	-47932	-
	45705	.45910		F119F.	.46318	-46522	.46726	.46930	.47133	.47336	.47538	01174	-47942	
	12684	.45889		.46094	-16298	-46502	40705	.46910	47113	-47315	47517	. 05115.	2567¥	
	*45664	.45869		.46073	11691.	.46481	.46685	.46889	-47092	-47205	16111	-47699	.47901	
	.00300	-00200		+00200+	-00500-	-00201	.00500	-00203	.00200	-00200	-00200	-00500-	.00200	
	-45643	.45848		-46058	16257	19191-	40065	69394.	-47072	.47275	LLF1F.	47679	147881	
	208	500		210	115	212	213	214	215	216	217	218	219	

### INDEX.

Adiabatic Expansion Curve	32
Adiabatic Expansion Condensation	35
Adiabatic Expansion of Wet Steam	37
Air Calculations, Gas Engine	83
Air Engines, Diagrams for	104
Aquene Curve	8
Areas of Diagrams, Comparison of	46
21 on of Programme, comparison of the comparison	
Balance Sheet, Heat, Steam Engine	72
Balance Sheet, Heat, Gas Engine	96
Boulvin's (Professor) Complete Entropy Diagram	28
Carnot Cycle	. 34
Chart, Theta-Phi	2
Clearance Volumes	21
Clearance Surfaces	54
Coefficient of Performance.	106
Compari-on of Areas	46
	28
Complete Entropy Diagram	23
Compound Engine, Diagram for	59
Compounding, Effect of	59
Condensation Coefficient	36
Condensation during Adiabatic Expansion	44
Condensation during Expansion	
Condensation, Initial	
Constant Volume Curves	92
Constant Temperature Lines	
Conversion of Indicator Diagram	21
Cotterill, Professor, on Steam Engine	
Critical Temperature	14
Cut-off, Most Economical	61
Density of Steam	109
Diesel Oil Motor, Professor Schröter's Test	97
Diesel Oil Motor, Temperature Calculations	98
Diesel Oll Motor Entrony Diagram	99
Diesel Oll Motor, Entropy Diagram	. 57
Donkin, Mr. B., on "Gas, Air, and Oil Engines"	81
Dryness Fraction, Calculation	21
Dryness Fraction, Comparisons of	56
Dighess Fraction, Companisons of	00
Efficiency, Thermal, Steam Engine	69
Efficiency, Thermal, Stirling's Air Engine	103
Efficiency, Thermal, Refrigerators 106,	108
Englue, Compound, Diagrams for	23
Engine, Single-cylinder, Diagrams for	46
Engline, Triple-expansion, Diagrams for	39

#### INDEX.

	100
Entropy	3
Entropy of Water	10
Entropy of Steam	11
Entropy of Superheated Steam	46
Entropy of Gases	74
Entropy Diagram	5
Entropy Diagram for Ice, Water, and Steam	11
Entropy Diagram for Gas Engine	87
Entropy Diagram for ()   Engine	07
Entropy Diagram for Oil Engine	104
Enjagen's Air Ungino	104
Ericsson's Air Engine. 1 Exchanges of Heat	104
Ewing, Professor, on Mechanical Production of Cold	LUO
Exchanges of Heat	08
Exhaust Period, Gas Engine	91
Exhaust Products, Gas Engine Exhaust Waste, Gas Engine	85
Exhaust Waste, Gas Engine	94
Expansion, Adiabatic 32, Expansion, Most Economical 32,	35
Expansion, Most Economical	61
Expansion Period, Gas Engine	90
Con Cool Analysis	0.4
Gas, Coal, Analysis	84
Gas, Coal, Specific ileat	84
Gases, Entropy of	73
Gases, Specific Heat of Gas Engine, Theoretical Gas Eugine, Temperature Calculations	73
Gas Engine, Theoretical	76
Gas Engine, Temperature Calculations	78
Gas Engine, Actual, 7 Horse Power	81
Gas Engine, Ideal Diagram	87
Gue Eugine Corrected Diagram	89
Gas Eugine, Corrected Diagram Gray, Mr. M. F., on "Theta-Phi Chart"	1
Gray, art. ar. F., Oil Indea-1 in Chart	A
Heat Balance Sheet, Steam Engine	72
Heat Bajance Sheet, Gas Engine.	96
Heat, Latent	6
Heat Losses, Steam Engine	39
Heat-Losses, Measurement of	98
Heat Losses Gas Engine	94
Heat Losses, Gas Engine Heat Recovery Lines	66
Heat Weight	
ricat weight	3
High-pressure Cylinder, $\theta \phi$ Diagram	25
Indicator Diagrams Compared with $ heta \phi$	3
Indicator Diagrams Converted to $ heta  \phi$	
Indicator Diagrams Converted to 9	21
Initial Condensation	63
Introduction of $ heta \phi$ Diagrams	1
Jacketing, Steam, Effect of	39
Jacketing, Steam, Effect of	39
Latent Heat	6
Logarithmic Curves	79
Logarithmic Curves.  Longridge Mr. M., on "Trials of a Compound Engine"  Losses of Heat.  59, 68,	61
Losses of Heat	94
the same data to the same and t	
Low-pressure Cylinder, $\theta \phi$ Diagram	26
Main Valve, Passage in	27
Mixture in Gas-engine Cylinder	82
	-
Oil Engines, $\theta \phi$ Diagram for	07
On Engines, v y Diagram for	97
Priming Water	63

	PAGE
Quality Curves	21
Radiation, Gas-engine	94
Re-evaporation during Expansion	44
Refrigerators, Closed Cycle	10
Refrigerators, Efficiency 106, Refrigerators, Open Cycle	108
Regenerator, Effect of	10:
Regenerator, Effect of	49
Sankey, Captain, on "Marine-engine Trials"	2
Scales for Entropy Diagram	97
Scales for Entropy Diagram Schröter, Prof., on "Diesel Oil Motor" Specific Heat, Water Specific Heat, Superheated Steam	10
Specific Heat, Superheated Steam	40
Specific Heat, Coal Gas	84
Specific Heat, Gas-engine Mixture. Specific Volume, Steam	86
Speed, Effect of	56
Standard Engine of Comparison	69
State Points	35
Steam, Entropy Diagram for Steam Jacketing, Effect of	, 11
Steam, Superheated, $\theta$ $\phi$ Diagram	
Steam, Superheated, Effect of	49
Steam, Wet, Expansion of	37
Stiriing's Air Engine	101
Superheated Steam, $\theta \phi$ Diagram for	46
Superheated Steam, Entropy of	48
Superheated Steam, Effect of Surfaces, Ciearance	49 54
Sullaces, Clearance	0,
Temperature of Combustion	94
Temperature, Constant, Lines	30
Temperature, Critical	
Temperature in Gas Engine	98
Theoretical Gas Engine	
Theoretical Temperature, Gas Engine	94
Thermal Efficiency, Steam Engine	, 69
Thermal Efficiency of Carnot Cycle Thermal Efficiency of Rankine Cycle.	69
Thermal Efficiency of Stirling's Air Engine	103
Thermai Efficiency of Refrigerators	108
Triple-expansion Engine' $\theta \phi$ Diagram	39
Volume Factor	, 45
Volume, Specific, Steam	17
Wall Action, Steam Engine	52
Wail Action, Gas Engine	
Water, Entropy of	. 10
Water, Specific Heat of	10
Weight of Steam 22, Wet Steam, Expansion of Williams, Mr. P. W., on "Non-condensing Steam-engine Trials"	37
Wilians, Mr. P. W., on "Non-condensing Steam-engine Trials"	2
Willans, Mr. P. W., on "Condensing Steam-engine Trials" 58	, 70
Work Losses	44

Second Edition.

Crown 8vo, cloth, price 4s. 6d. net, post free anywhere.

## THE INDICATOR AND ITS DIAGRAMS: WITH CHAPTERS ON ENGINE AND BOILER TESTING.

By CHARLES DAY, Wh.Sc. Including a Table of Piston Constants,

Crown 8vo, cloth, price 3s. 6d. net, post free anywhere.

DROBLEMS IN MACHINE DESIGN. For the Use of Students, Draughtsmen, and others. By CHAS. H. INNES, M.A., Lecturer on Engineering at the Rutherford College, Newcastleon-Tyne.

Crown 8vo, cloth, price 6s. net, post free anywhere.

THE APPLICATION OF GRAPHIC METHODS TO THE DESIGN OF STRUCTURES. Specially prepared for the use of Engineers. Profusely illustrated. By W. W. F. PULLEN, Wh.Sc., Assoc.M.Inst.C.E., M.I.Mech.E.

Crown 8vo, price 3s. 6d. net, post free anywhere.

PENTRIFUGAL PUMPS, TURBINES, & WATER MOTORS: including the Theory and Practice of Hydraulics (specially adapted for engineers). By CHAS. H. INNES, M.A., Lecturer on Engineering at Rutherford College, Newcastle-on-Tyne.

Crown 8vo, cloth, price 3s. net, post free anywhere.

ENGINEERING ESTIMATES AND COST ACCOUNTS. By F. G. BURTON, formerly Secretary and General Manager of the Milford Haven Shipbuilding and Engineering Co. Limited.

Crown 8vo, price 2s. net, post free anywhere.

PENING BRIDGES. By GEORGE WILSON, M.Sc., Demonstrator and Assistant Lecturer at the University College of South Wales, Cardiff.

Crown 8vo, cloth, price 2s. 6d. net, post free anywhere.

THE NAVAL ENGINEER AND THE COMMAND OF THE SEA. A Story of Naval Administration. By FRANCIS G. BURTON, author of "Engineering Estimates and Cost Accounts," &c.

Crown 8vo, cloth, price 3s. 6d. net, post free anywhere.

# RAPHIC METHODS OF ENGINE DESIGN. By A. H. BARKER, B.A., B.Sc., Wh.Sc., author of "Graphical

Calculus," &c.

Crown 8vo, cloth, price 3s. 6d. net, post free anywhere.

INJECTORS: THEORY, CONSTRUCTION, AND WORKING. By W. W. F. PULLEN, Wh.Sc., Assoc.M.Inst.C.E., M.Inst.M.E.

Crown 8vo, cloth, price 2s. 6d. net, post free anywhere.

PRACTICAL NOTES ON THE CONSTRUCTION OF CRANES AND LIFTING MACHINERY. EDWARD C, R. MARKS, Assoc. M, Inst. C.E., M, I, Mech. E.

Crown 8vo, cloth, price 4s. 6d. net, post free anywhere.

MODERN GAS AND OIL ENGINES. Profusely illustrated. A full and exhaustive Treatment of the Design, Construction, and Working of Gas and Oil Engines up to date. By FREDERICK GROVER, Assoc.M.Inst.C.E.

Crown 8vo, cloth, price 6s. net, post free anywhere.

HEAT AND HEAT ENGINES. A Treatise on Thermodynamics as Practically Applied to Heat Motors. Specially written for Engineers. By W. C. POPPLEWELL, M.Sc.

Crown 8vo, cloth, price 5s. net, post free anywhere.

MARINE ENGINEERS: THEIR QUALIFICA-TIONS AND DUTIES. With Notes on the Care and Management of Marine Engines, Boilers, Machinery, &c. By E. G. CONSTANTINE, Assoc.M.Inst.C.E., M.I.Mech.E.

Crown 8vo, cloth, price 3s. net, post free anywhere.

THE ABC of the DIFFERENTIAL CALCULUS.

By W. D. WANSBROUGH, author of "Portable Engines," "Proportions and Movement of Slide Valves," &c.

THE TECHNICAL PUBLISHING CO. LIMITED, 31, WHITWORTH STREET, MANCHESTER, ENGLAND; JOHN HEYWOOD, LONDON AND MANCHESTER; And all Booksellers.

